Date: September 1st, 1957

Aircraft: C-105 MK 2

Production A/C

<u>S E C R E T</u> Report # 7-0400-34 Iss. 11

INDEX

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| 1-1 to 1-3 | Introductory notes and brief explanation of weight changes. |
| 2-1 to 2-2 | Detailed explanation of weight changes to weight reports as listed on I.B.M. Sheets. |
| 3-1 to 3-2 | Weight & C.G. Summaries. |
| 4-1 to 4-15 | I.B.M. Detail Sheets of Weights and C.G?s. |

N. B. The flight C.G. envelopes for the MK 2 Aircraft have been omitted until such time as a fully approved fuel sequencing is established.

Classification cancelled / changed to: <u>UNCLASSSIFIED</u>
By authority of: <u>DRDA 7/DARFT 5-8/DAS Eng 6-4-5</u>

Date: 5 Nov 1992

Signature: Duulee,

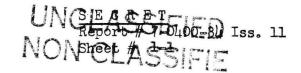
Unit / Rank / Appointment DSIS 3, Secretary CRAD HQ DRP





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INTRODUCTION

The following is a Weight & C.G. Summary of the C-105 MK 2 Production Aircraft, based on the latest weight estimates available.

All Weight and C.G. changes are relative to Issue 10 of August 1st, 1957.

GENERAL:

- a) Orenda PS 13 Engines comprise the Power Plant (4,500 lb each excluding Nose Bullet and Input Frame).
- b) A package containing 4 "semi-submerged" Sparrow II Missiles (432 lb each) forms the current armament.
- c) The R.C.A. Radar Astra I System is installed, weight & C.G. location being to the latest breakdown received from them, dated 3rd July 1957. The Sparrow III Auxiliaries, included in the breakdown, have been deleted, however, the Infra Red Tracker System has been re-instated as a requirement; some Avro installed Antennae have been added. (Total Weight = 2,741.0 lb).

In addition to above the Minneapolis-Honeywell MH & Damping System is installed.

The early MK 2 A/C will have the "Astra Minus" System installed (i.e. navigational and communication equipment only)

d) A considerable number of MK l parts, assemblies and equipment have now been weighed. Where these weights apply to the MK 2 Aircraft, they have been recorded in the MK 2 records.

On the I.B.M. Sheets in this report, immediately preceeding the item title, will be found a number varying from 0 to 100. This is the percentage actual weight recorded in that report.

N. B. It should be noted that the early MK 2 Aircraft will not be as this summary designates, but will be flight test versions with instrumentation replacing the missile package etc. or for development of Astra I Radar System. etc.

Ref. 2-1

1. STRUCTURE

WEIGHT (1b)

a) Wing:

Inner Wing - production drawing estimates of skins and centre trailing edge structure

Total Wing Decrease

- 25

b) Fin & Rudder

No Weight Change.

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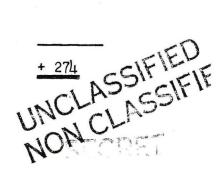
INTRODUCTION

| STR | UCTURE | I CH' | r (1b) |
|------------|---|-------|---------------|
| c) | Fuselage fwd. Sta. 255" | | |
| | Radome - due to unsatisfactory electrical properties solid laminate will replace the filled honey-comb laminate Air Intakes - estimate of MK 2 design Fwd. Fuselage Increase Ref. 2-1 | + = + | 75 3 72 |
| d) | Centre Fuselage Sta. 255"-485" | | |
| | Production drawing estimates of Formers, armament bay roof and C.F. Ducts Centre Fuselage Decrease Ref. 2-1 | 9 | 15 15 |
| e) | Duct Bay Sta. 485"-591.65" | | |
| | Floating Duct - estimated to preliminary schemes Joint items & H/Exchanger Duct included here (were in Engine Bay and Engine Group resp.) Total Duct Bay Increase Ref. 2-1 | +++ | 18 65 |
| f) | Engine Bay Sta. 591.65"-742.5" | | |
| | Engine Tunnel - increase in insulation based on MK LA/C etc Joint Engine to Duct - deletion of allowance (see above) Total Engine Bay Decrease Ref. 2-2 | | 6 12 6 |
| g) | Rear Fuselage Sta. 742.5" Aft. | | |
| | Fixed section 742.5"-803" - first estimates of formers etc Stinger & Tailcones - N155 cobalt based alloy to be used with Titanium O/Skin on Tailcones, previously weight recorded was for an entirely Titanium structure Total Rear Fuselage Increase Ref. 2-2 | + ; | 168 |
| h) | Fuselage Joints | | |

Fuserage Joints

No Weight Change.

TOTAL STRUCTURAL INCREASE



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INTRODUCTION

| 2. | LANDING GEAR | WEIGHT (1b) |
|------|--|--------------------------------------|
| | No Weight Change | |
| 3. | POWER PLANT & SERVICES | |
| | Engines - Orenda's earlier estimate did not include inlet frames H/Exchanger Duct with floating duct (see above) Fire Extinguishing System - Actual Weight of bottles Fuel System - Miscellaneous design changes TOTAL POWER PLANT & SERVICES INCREASE | + 160 = 6 = 5 + 23 + 172 |
|), , | Ref. 2-2 | |

4. FLYING CONTROLS GROUP

No Weight Change

Issue 10

5. EQUIPMENT GROUP

Radar System: - Revised Weights from R.C.A. The I.R. Tracker System and Radar Homer Antenna, previously deleted from this report, have now been reinstated as basic requirements + 127 TOTAL EQUIPMENT GROUP INCREASE Ref. 2-2

SUMMARY

Weight Change - Aircraft Basic Weight

| Structure | + | 274 | lb。 |
|-------------|---|-----|-----|
| Power Plant | + | 172 | lb. |
| | + | 127 | lb. |
| | + | 573 | lb. |

Weight Change - Operational Weight Empty (A/C less fuel)

Issue 11

| | Charles and the Control of the Contr | |
|----------|--|--------|
| Щ,516 lb | 45,089 lb = | 573 lb |



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DETAILED WEIGHT CHANGES TO I.B.M. TABULATED DETAILS OF AUG. 1st, 1957

WEIGHT CHANGES

WEIGHT (1b)

Wings:

- 36.70 I/W Skins - now estimated to production drawings, an error was made in the early estimates

+ 11.71 I/W Centre Trailing Edge - production drawing estimates increases in caps and webs of beams, particularly # 4 and # 5 for rear engine mountings.

- 24.99

Front Fuselage Fwd. 255"

+ 75.50 Radome - solid laminate replaces sandwich construction the electrical properties of the lighter sandwich type are unsatisfactory.

- 3.26 Air Intakes - estimated to production drawings for MK 2 design

+ 72.24

Centre Fuselage Sta. 255" - 485"

+ 9.63 Formers C.F. - production drawing estimates incorporating minor changes (approx. + 0.5 lb/former)

- 6.94 Armament Bay Roof - production drawing estimates some design changes.

- 17.34 Duct C.F. - alterations to allow for ram air duct doors (doors and mechanism are included with Air Conditioning see Equip. Group)

- 14.65

Duct Bay Sta. 485"-591.65"

+ 65.00 Floating Duct - estimated to preliminary scheme drawings - previously only allowances

Addition of Joint items (see

Engine Bay

Addition of H/Exch. Duct (see

Engine Group.

+ 46.75

+ 12.00

+ 65.00

+ 65,00

Engine Bay Sta. 591.65 - 742.5

+ 6.02 Engine Tunnel - increase in insulation weight based on tolerances of actual weights of MK 1 insulation

Addition of floating duct support structure



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DETAILED WEIGHT CHANGES TO I.B.M. TABULATED DETAILS OF AUG. 1st, 1957

WEIGHT CHANGES

WEIGHT (1b)

Engine Bay Sta. 591.65"-742.5" (cont d.)

- 12.00 Joint Engine to Duct - cancelled, now included with floating Duct (see Duct Bay above)
- 5.98

Rear Fuselage Sta. 742.5" Aft.

+ 13.91 Formers Fixed R.F. - first estimates to MK 2 drawings

+ 0.84 Engine Doors - two hinge pins previously omitted

+ 15.27 Parachute Box & Stinger - redesigned, stinger now in N155, Cobalt based alloy, was previously recorded in Titanium.

+ 153.15 Tailcones - first estimate to MK 2 lines using preliminary gauges supplied by Stress Office,
structure of N155 except O/Skin of Titanium.
Possible further increases may be expected
since proposed gauges tend to be too light.

+ 183,17

Power Plant & Services

| 3 | 5.00 | Fire Extinguishing System - Actual weight of bottles | | |
|---|--------|--|------|--------|
| | | as in MK 1 A/C | | |
| + | 154.00 | Engines PS13 - Engine C.G. moved forward by 2.00" | | |
| | | Inlet frames were not included in | | |
| | | Orenda s previous estimate. | + | 160.00 |
| | | Heat Exchanger Duct now with | | |
| | | Floating Duct. (see Duct Bay) | (30) | 6.00 |
| | | , | + . | 154.00 |
| + | 23.13 | Fuel System - filter added between heat exchanger | | |
| | | and Engine to reduce risk of Engine | | |
| | | contamination | + | 9.00 |
| | | Fuel no-air valves Actual weight | | |
| | | ■ 3.5 1b manufacturer s weight was | | |
| | | 3.0 lb (22 off) | + | 11.00 |
| | | Miscellaneous other changes | + | 3.13 |
| | | | + | 23.13 |

+ 172.13

Equipment Fixed & Removable.

| + | 10.00 | Radar | fixed | 0 | Homer | Antenna | re-instated |
|---|-------|-------|-------|---|-------|---------|-------------|
| | | | | | | | |

+ 61.50 Radar Remov. = latest R.C.A. estimates, I.R. Tracker system re-inst. (70 lb)

+ 55.00 Missile Pack Electronics - latest R.C.A. estimate for Sparrow II auxiliaries (Sparrow III auxiliaries are not carried)

+ 126.50

+ 573.42 TOTAL WEIGHT CHANGES



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WEIGHT & C.G. SUMMARY

| | WEIGHT | H. ARM | V. ARM. |
|--|-----------|---------|-----------|
| DEGCETON | | 1 | |
| DESCRIPTION | 1bo | ins. | ins. |
| STRUCTURE | 18,876,21 | 564.52 | 137.43 |
| Wings | 10,003.73 | 642.74 | 142.23 |
| Fin & Rudder | 1,024.00 | 754.25 | 209.28 |
| | 2,565.86 | 182.44 | 128.41 |
| Fuselage: Fwd. Sta. 255" Sta. 255" - 485" | 1,690.10 | 374.96 | 130.07 |
| Sta. 485" = 591.65" | 1,110,19 | 538.21 | 105.40 |
| | 1,547.28 | 661.11 | 110.57 |
| Sta. 591.65" - 742.5" | 882.72 | 801.05 | |
| Sta. 742.5" Aft. | | | 129.40 |
| "Marry-up" | 52.33 | 467.60 | 103.84 |
| UNDERCARRIAGE - Retracted | 2,551.77 | 487'.22 | 134.79 |
| Main Undercarriages | 1,901.62 | 539.31 | 141.00 |
| Main U/C Doors & Fairings | 291.80 | 537.10 | 138.40 |
| Nose Undercarriage | 333.81 | 170.80 | 99.70 |
| Nose U/C Door & Fairing | 24.54 | 162.24 | 88.23 |
| POWER PLANT & SERVICES | 10,800.67 | 671.41 | 121.32 |
| Engines & Accessories PS 13 | 9,186.78 | 687.95 | 121.16 |
| Gear Box & Drives on Fuselage | 281.84 | 601.70 | 102.49 |
| Engine Controls | 32.43 | 375.76 | 118.61 |
| Gear Box, Starter & Drives on Engine | 315.45 | 615.98 | 105.24 |
| Engine Nose Bullet (Orenda Supplied.) | 70.00 | 587.17 | 116.00 |
| Fire Extinguishing System | 65.46 | 700.45 | 134.21 |
| Engine Mountings | 132,38 | 666.82 | 136.52 |
| Fuel System | 716.33 | 530.91 | 134.48 |
| FLYING CONTROLS GROUP | 1,780.64 | 685.55 | 140.51 |
| Mechanical Flying Controls | 946.48 | 687.84 | 148.61 |
| Hydraulic Flying Controls | 834.16 | 682.95 | 131.31 |
| EQUIPMENT FIXED & REMOVABLE | 8,229.95 | 328.80 | 112.39 |
| Instruments | 46.07 | 163.68 | 138.70 |
| Probe | 15.00 | -38.14 | 108.00 |
| Cockpit Pressure Sealing | 5.00 | 186.00 | 130.00 |
| Oxygen System | 23.59 | 253.72 | 156.43 |
| Ejector Seats | 284.42 | 202.80 | 136.25 |
| Air Conditioning System | 856.00 | 333.49 | 134.98 |
| Hydraulic Main System | 623.15 | 502.78 | 117 - 75 |
| Cahin Insulation | 14.31 | 187.48 | 132.00 |
| Brake Parachute | 62.38 | 769.41 | 143.24 |
| Electrical System | 1,259.16 | 432.22 | 112.68 |
| Low Pressure Pneumatics | 55.70 | 416.79 | 123.99 |
| Surface Finish | 100,00 | 591.52 | 140.20 |
| Intake De-icing Boots | 51.84 | 197.02 | 118.00 |
| Radome Anti-icing | 8.88 | 51.49 | 125.00 |
| Canopy Actuation | 64.92 | 221.99 | 154.35 |
| Cabin Consoles | 17.45 | 174.76 | 124.34 |
| Radar Door Actuation | 10.00 | 268.00 | 95.00 |
| M.H. 64 Damping System | 139.13 | 449.26 | 135.57 |
| Radio & Radar Removable | 1,912.00 | 180.31 | 103.97. |
| Radio & Radar Fixed | 631.00 | | |
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WEIGHT & C.G. SUMMARY

| DESCRIPTION | - | WEIGHT 1b | H. ARM ins. | V. ARM ins. | M.A.C. |
|------------------------------------|--------|---------------|----------------|----------------|---|
| Equipment (Fixed & Removable) (con | tid.) | | | | |
| | | 0 | | - (| |
| Sparrow Pack Structure | | 870.70 | 389.60 | 96.00 | |
| Sparrow Pack Mechanisms | | 582.57 | 379.30 | 99.04 | |
| Sparrow Pack Hydraulics | | 331.94 | 373.52 | 99.00 | |
| Sparrow Pack Electronics | | 198.00 | 327.96 | 100.00 | |
| Sparrow Pack Electrics | | 64.10 | 362.29 | 95.00 | |
| Sparrow Pack Air Conditioning | | 2.64 | 309.93 | 104.15 | |
| AIRCRAFT BASIC WEIGHT | | 42,239.24 | 546.36 | 128.40 | |
| USEFUL LOAD (less fuel) | | 2,850.76 | 372.91 | 105.43 | |
| Crew | | 430.00 | 194.00 | 136.50 | |
| Oil | | 138.97 | 636.92 | 110.57 | |
| Alcohol | | 22.00 | 93.00 | 138.00 | |
| Engine Fire Extinguishing Fluid | | 25.00 | 730.00 | 129.00 | |
| Residual Fuel | | 218.40 | 553.98 | 134.04 | |
| Missiles | | 1,728.00 | 389.29 | 88.30 | |
| Oxygen Charge | | 13.39 | 259.68 | 159.91 | |
| Water for Air Conditioning | | 275.00 | 268.00 | 131.75 | |
| U/C | Up | | 535.40 | 126.95 | 27.46 |
| Operational Weight Empty | | 45,090.00 | 737.4 | | -104- |
| | Down | 429-7-0 | 537.52 | 123.44 | 28.05 |
| U/C | | | 541.22 | 128.49 | 29.07 |
| O.W.E. less Missiles | °P | 43,362,00 | >4-0 | | -, 0-1 |
| | Down | 4292-201 | 543.43 | 124.84 | 29.68 |
| Normal Combat Mission Fuel * | | | | | |
| (2,044 gals. @ 7.8 lb/gal.) | i i | 15,940.00 | 542.03 | 142.30 | |
| U/C | qU | | 537.13 | 130.96 | 27.94 |
| Normal Combat Weight | - | 61,030.00 | | | *************************************** |
| | Down | , - | 538.70 | 128.37 | 28.37 |
| Half Combat Mission Fuel | | | | | |
| (1,022 gals. @ 7.8 lb/gal.) | | 7,970.00 | 543.18 | 139.55 | - |
| U/C | Up | | 536.57 | 128.84 | 27.78 |
| Combat Weight (half | - | | | | |
| Mission Fuel | | 53,060.00 | 41 | | |
| | Down | | 538.37 | 125.86 | 28.28 |
| Maximum Internal Fuel (2,492 | | | | | |
| gals. @ 7.8 lb/gal.) | j | 19,438.00 | 541.85 | 144.16 | |
| U/C | qU | | 537.34 | 132.13 | 28.00 |
| A.U.W. Maximum Internal Fuel | - | 64.528.00 | | | |
| | Down | .,,, | 538.82 | 129.68 | 28.41 |
| Maximum External Fuel (500 | | | | | |
| gals. @ 7.8 lb/gal. + Drop Tank | | 4,248.00 | 521.84 | 60.67 | |
| U/C | gU | 7,240,000 | 536.38 | 127.72 | 27.73 |
| A.U.W. Maximum Internal and | | 68,776,00 | // - 6/ | :014 | -1017 |
| | Down | y 1 1 - 0 - 3 | 537.77 | 125.42 | 28.12 |
| 0/0 | - 1144 | 1 | 721011 | | |

N.B. 1) * New figures are being prepared by Aerodynamics for Mission Fuel in accordance with latest PS 13 data, but these are not yet available.

²⁾ Aircraft Datum = 120" above and arbitrarily chosen ground