AVRO AIRCRAFT LIMITED

Inter-Departmental Memorandum

Ref 7659/05/J
Date March 28, 1958
To S. E. Harper
From J. D. Hodge
Subject ARROW 1 - FUEL SYSTEM TESTS

Herewith R.F.T. No. 5040, Fuel System Engineering Tests which lists the testing required to prove the functioning of the Arrow 1 Fuel System.

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MALTON, ONTARIO

R.F.T. NO.	5040			
SHEET NO.	<u> </u>	OF.		
DATE:	March	28,	1958	

REQUISITION FOR FLIGHT TEST

25201 AIRCRAFT 25202 and/or 25203	ASSIGNMENT NO. X73=384	WORK ORDER NO.

DATE:

FUEL SYSTEM ENGINEERING TESTS

OBJECT 1.

To prove the Arrow I fuel system on a complete aircraft.

INSTRUMENTATION REQUIRED 2.

The instrumentation required is as in section 4 of FAR/ClO5/1 and for convenience a copy of this section is attached.

PROCEDURE 30

It is required to record all fuel system parameters as listed in section 2 pproxduring the following manoeuvres

- 3.1 Take off and acceleration to climb speed.
 - 3.1.1 at military rating 3.1.2 with afterburner.
- 3.2 Climb from sea level, without the afterburner at M .92 to 35,000 ft and then at best climb speed to ceiling (without afterburner).
- 3.3 With the afterburner, climb at M .92 to 35,000 ft. and then at M 1.5 to 50,000 ft.
- 3.4 Climb to 50,000 ft, with the afterburner on and at the maximum rate of climb
- 3.5 Level flight at the aircraft operational ceiling.
- 3.6 Dive from 50,000 ft to 15,000 ft with the following starting conditions
 - 3.6.1 Starting at M 1.5 with the power on. 3.6.2 Starting at M 1.5 with the engines idling.

R.F.T., PREPARED BY:	APPROVED BY:	AUTHORIZED
DATE FOR COMPLETION	PRIORITY	ESTIMATED COMPLETION

MALTON, ONTARIO

R.F.T. NO.	5040		
SHEET NO.	2	QF	
DATE:	March	28, 1958	

REQUISITION FOR FLIGHT TEST

25201 AIRCRAFT 25202 and/or 25203	assignment no. X73-384	WORK ORDER NO.

3.7 To establish inverted flight endurance under the following conditions.

3.7.1 At 10,000 ft. with throttle set for cruise at M .92.

3.7.2 At 35,000 ft. with throttle set for cruise at M .92.

3.7.3 At 35,000 ft. with throttle set for cruise at M 1.5.

3.7.4 At 50,000 ft. with throttle set for cruise at M 1.5.

NOTE: The estimated inverted flight endurance is 15 seconds at S.L. and 45 seconds at 50,000 ft. and it is required to establish these times more accurately.

It is expected that achieving inverted flight by roll and by pitch may give different results, therefore it is requested that the pilot completes these tests by rolling into inverted flight and wherever possible to achieve the attitude by a half-loop as well.

4. DATA REQUIRED

Recordings from all the instrumentation are required throughout the specified manoeuvres. Also required are the pilots comments on the adequacy of the fuel quantity indicating system and the frequency and validity of all warning lights.

R.F.T. PREPARED BY:	APPROVED BY:	AUTHORIZED BY:
DATE FOR COMPLETION	PRIORITY	ESTIMATED COMPLETION DATE: