

QC
Avro
C-105
Rpt
7-0400-34
Iss-11

QCX
Avro
CF105
R-7-0400-34
Iss-11

FILE IN VAULT

C-105 MK 2 Report # 7-0400-34
Prod. A/C Issue 11

WEIGHT SUMMARY & C.G. POSITION 19

R.C.A.F.

Sept. 1/57

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TO
A

DATE
JAN 07 1993
Dec. 7, 1992

Report no.: QCX - AVRO - CF105- R-7-0400-34-155-11

has been ☐ downgraded to: _____

☒ de-classified

by (Name): Michel W. Drapeau

(Dept.): A/DND Coordinator, Access to Information

Date: Dec. 7, 1992

R. Auger
Signature

Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

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SECRET
Report # 7-0400-34 Iss. 11

I N D E X

<u>Sheet #</u>	<u>Content</u>
1-1 to 1-3	Introductory notes and brief explanation of weight changes.
2-1 to 2-2	Detailed explanation of weight changes to weight reports as listed on I.B.M. Sheets.
3-1 to 3-2	Weight & C.G. Summaries.
4-1 to 4-15	I.B.M. Detail Sheets of Weights and C.G's.

N. B. The flight C.G. envelopes for the MK 2 Aircraft have been omitted until such time as a fully approved fuel sequencing is established.

Classification cancelled / changed to: UNCLASSIFIED

By authority of: DRDA 7/DARFT 5-8/DAS Eng 6-4-5

Date: 5 Nov 1992

Signature: *Laubrey*

Unit / Rank / Appointment: DSIS 3, Secretary CRAD HQ DRP



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49133

Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

UNCLASSIFIED
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Report # 7-0400-34 Iss. 11
Sheet # 1-1
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INTRODUCTION

The following is a Weight & C.G. Summary of the C-105 MK 2 Production Aircraft, based on the latest weight estimates available.

All Weight and C.G. changes are relative to Issue 10 of August 1st, 1957.

GENERAL:

- a) Orenda PS 13 Engines comprise the Power Plant (4,500 lb each excluding Nose Bullet and Input Frame).
- b) A package containing 4 "semi-submerged" Sparrow II Miissiles (432 lb each) forms the current armament.
- c) The R.C.A. Radar Astra I System is installed, weight & C.G. location being to the latest breakdown received from them, dated 3rd July 1957. The Sparrow III Auxiliaries, included in the breakdown, have been deleted, however, the Infra Red Tracker System has been re-instated as a requirement; some Avro installed Antennae have been added. (Total Weight = 2,741.0 lb).

In addition to above the Minneapolis-Honeywell MH 64 Damping System is installed.

The early MK 2 A/C will have the "Astra Minus" System installed (i.e. navigational and communication equipment only)

- d) A considerable number of MK 1 parts, assemblies and equipment have now been weighed. Where these weights apply to the MK 2 Aircraft, they have been recorded in the MK 2 records.

On the I.B.M. Sheets in this report, immediately preceeding the item title, will be found a number varying from 0 to 100. This is the percentage actual weight recorded in that report.

N. B. It should be noted that the early MK 2 Aircraft will not be as this summary designates, but will be flight test versions with instrumentation replacing the missile package etc. or for development of Astra I Radar System. etc.

1. STRUCTURE

WEIGHT (lb)

a) Wing:

Inner Wing - production drawing estimates of skins and
centre trailing edge structure

Total Wing Decrease

Ref. 2-1

25
25

b) Fin & Rudder

No Weight Change.

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Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

SECRET
Report # 7-0400-34-11
Sheet # 1-2
NON CLASSIFIED

INTRODUCTION

1. STRUCTURE

WEIGHT (lb)

c) Fuselage fwd. Sta. 255"

Radome - due to unsatisfactory electrical properties
solid laminate will replace the filled honey-
comb laminate

+ 75

Air Intakes - estimate of MK 2 design

= 3

Fwd. Fuselage Increase

+ 72

Ref. 2-1

d) Centre Fuselage Sta. 255"-485"

Production drawing estimates of Formers, armament bay
roof and C.F. Ducts

= 15

Centre Fuselage Decrease

= 15

Ref. 2-1

e) Duct Bay Sta. 485"-591.65"

Floating Duct - estimated to preliminary schemes

+ 47

Joint items & H/Exchanger Duct

included here (were in Engine

Bay and Engine Group resp.)

+ 18

Total Duct Bay Increase

+ 65

Ref. 2-1

f) Engine Bay Sta. 591.65"-742.5"

Engine Tunnel - increase in insulation based on MK 1A/C etc + 6

Joint Engine to Duct - deletion of allowance (see above) = 12

Total Engine Bay Decrease

= 6

Ref. 2-2

g) Rear Fuselage Sta. 742.5" Aft.

Fixed section 742.5"-803" - first estimates of formers etc + 15

Stinger & Tailcones - N155 cobalt based alloy to be used

with Titanium O/Skin on Tailcones,

previously weight recorded was for

an entirely Titanium structure

+ 168

Total Rear Fuselage Increase + 183

Ref. 2-2

h) Fuselage Joints

No Weight Change.

TOTAL STRUCTURAL INCREASE

+ 274

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Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

UNCLASSIFIED
SECRET
Report # 7-0400-34 Iss. 11
Sheet # 1-3
NON CLASSIFIE

INTRODUCTION

2. LANDING GEAR

WEIGHT (lb)

No Weight Change

3. POWER PLANT & SERVICES

Engines - Orenda's earlier estimate did not include inlet frames	+ 160
H/Exchanger Duct with floating duct (see above)	- 6
Fire Extinguishing System - Actual Weight of bottles	- 5
Fuel System - Miscellaneous design changes	+ 23
<u>TOTAL POWER PLANT & SERVICES INCREASE</u>	<u>+ 172</u>

Ref. 2-2

4. FLYING CONTROLS GROUP

No Weight Change

5. EQUIPMENT GROUP

Radar System: - Revised Weights from R.C.A. The I.R. Tracker System and Radar Homer Antenna, previously deleted from this report, have now been re-instated as basic requirements	+ 127
<u>TOTAL EQUIPMENT GROUP INCREASE</u>	<u>+ 127</u>

Ref. 2-2

SUMMARY

Weight Change - Aircraft Basic Weight

Structure	+ 274 lb.
Power Plant	+ 172 lb.
Equipment	+ 127 lb.
	<u>+ 573 lb.</u>

Weight Change - Operational Weight Empty (A/C less fuel)

<u>Issue 10</u>	<u>Issue 11</u>	
<u>44,516 lb</u>	<u>45,089 lb</u>	= <u>573 lb</u>

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Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

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SECRET
Report # 7-0400-34 Iss 11
Sheet # 2-1
NON CLASSIFIE

DETAILED WEIGHT CHANGES TO I.B.M.
TABULATED DETAILS OF AUG. 1st, 1957

WEIGHT CHANGES

WEIGHT (lb)

Wings:

- 36.70 I/W Skins - now estimated to production drawings, an error was made in the early estimates
- + 11.71 I/W Centre Trailing Edge - production drawing estimates increases in caps and webs of beams, particularly # 4 and # 5 for rear engine mountings.

- 24.99

Front Fuselage Fwd. 255"

- + 75.50 Radome - solid laminate replaces sandwich construction the electrical properties of the lighter sandwich type are unsatisfactory.
- 3.26 Air Intakes - estimated to production drawings for MK 2 design

+ 72.24

Centre Fuselage Sta. 255" - 485"

- + 9.63 Formers C.F. - production drawing estimates incorporating minor changes (approx. + 0.5 lb/former)
- 6.94 Armament Bay Roof - production drawing estimates some design changes.
- 17.34 Duct C.F. - alterations to allow for ram air duct doors (doors and mechanism are included with Air Conditioning see Equip. Group)

- 14.65

Duct Bay Sta. 485"-591.65"

- + 65.00 Floating Duct - estimated to preliminary scheme drawings - previously only allowances
- + 46.75
- + 12.00
- + 6.25
- + 65.00

+ 65.00

Engine Bay Sta. 591.65 - 742.5

- + 6.02 Engine Tunnel - increase in insulation weight based on tolerances of actual weights of MK 1 insulation
- + 3.38
- + 2.64
- + 6.02

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Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

UNCLASSIFIED
SECRET
Report # 7-0400-34 Iss. 11
Sheet # 2-2
NON CLASSIFIE

DETAILED WEIGHT CHANGES TO I.B.M.
TABULATED DETAILS OF AUG. 1st, 1957

WEIGHT CHANGES

WEIGHT (lb)

Engine Bay Sta. 591.65"-742.5" (cont'd.)

- 12.00 Joint Engine to Duct - cancelled, now included with floating Duct (see Duct Bay above)
- 5.98

Rear Fuselage Sta. 742.5" Aft.

- + 13.91 Formers Fixed R.F. - first estimates to MK 2 drawings
- + 0.84 Engine Doors - two hinge pins previously omitted
- + 15.27 Parachute Box & Stinger - redesigned, stinger now in N155, Cobalt based alloy, was previously recorded in Titanium.
- + 153.15 Tailcones - first estimate to MK 2 lines using preliminary gauges supplied by Stress Office, structure of N155 except O/Skin of Titanium. Possible further increases may be expected since proposed gauges tend to be too light.

+ 183.17

Power Plant & Services

- 5.00 Fire Extinguishing System - Actual weight of bottles as in MK 1 A/C
- + 154.00 Engines PS13 - Engine C.G. moved forward by 2.00"
Inlet frames were not included in Orenda's previous estimate. + 160.00
Heat Exchanger Duct now with
Floating Duct. (see Duct Bay) = 6.00
+ 154.00
- + 23.13 Fuel System - filter added between heat exchanger and Engine to reduce risk of Engine contamination + 9.00
Fuel no-air valves Actual weight
= 3.5 lb manufacturer's weight was
3.0 lb (22 off) + 11.00
Miscellaneous other changes + 3.13
+ 23.13

+ 172.13

Equipment Fixed & Removable.

- + 10.00 Radar fixed - Homer Antenna re-instated
- + 61.50 Radar Remov. - latest R.C.A. estimates, I.R. Tracker system re-inst. (70 lb)
- + 55.00 Missile Pack Electronics - latest R.C.A. estimate for Sparrow II auxiliaries (Sparrow III auxiliaries are not carried)

+ 126.50

+ 573.42 TOTAL WEIGHT CHANGES

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Date: September 1st, 1957
Aircraft: C-105 MK2
Production A/C

SECRET
Report # 7-0100-34 Iss. 11
Sheet # 3-1
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WEIGHT & C.G. SUMMARY

DESCRIPTION	WEIGHT lb.	H. ARM ins.	V. ARM. ins.
STRUCTURE	18,876.21	564.52	137.43
Wings	10,003.73	642.74	142.23
Fin & Rudder	1,024.00	754.25	209.28
Fuselage: Fwd. Sta. 255"	2,565.86	182.44	128.41
Sta. 255" - 485"	1,690.10	374.96	130.07
Sta. 485" - 591.65"	1,110.19	538.21	105.40
Sta. 591.65" - 742.5"	1,547.28	661.11	110.57
Sta. 742.5" Aft.	882.72	801.05	129.40
"Marry-up"	52.33	467.60	103.84
UNDERCARRIAGE - Retracted	2,551.77	487.22	134.79
Main Undercarriages	1,901.62	539.31	141.00
Main U/C Doors & Fairings	291.80	537.10	138.40
Nose Undercarriage	333.81	170.80	99.70
Nose U/C Door & Fairing	24.54	162.24	88.23
POWER PLANT & SERVICES	10,800.67	671.41	121.32
Engines & Accessories PS 13	9,186.78	687.95	121.16
Gear Box & Drives on Fuselage	281.84	601.70	102.49
Engine Controls	32.43	375.76	118.61
Gear Box, Starter & Drives on Engine	315.45	615.98	105.24
Engine Nose Bullet (Orenda Supplied.)	70.00	587.17	116.00
Fire Extinguishing System	65.46	700.45	134.21
Engine Mountings	132.38	666.82	136.52
Fuel System	716.33	530.91	134.48
FLYING CONTROLS GROUP	1,780.64	685.55	140.51
Mechanical Flying Controls	946.48	687.84	148.61
Hydraulic Flying Controls	834.16	682.95	131.31
EQUIPMENT FIXED & REMOVABLE	8,229.95	328.80	112.39
Instruments	46.07	163.68	138.70
Probe	15.00	38.14	108.00
Cockpit Pressure Sealing	5.00	186.00	130.00
Oxygen System	23.59	253.72	156.43
Ejector Seats	284.42	202.80	136.25
Air Conditioning System	856.00	333.49	134.98
Hydraulic Main System	623.15	502.78	117.75
Cabin Insulation	14.31	187.48	132.00
Brake Parachute	62.38	769.41	143.24
Electrical System	1,259.16	432.22	112.68
Low Pressure Pneumatics	55.70	416.79	123.99
Surface Finish	100.00	591.52	140.20
Intake De-icing Boots	51.84	197.02	118.00
Radome Anti-icing	8.88	51.49	125.00
Canopy Actuation	64.92	221.99	154.35
Cabin Consoles	17.45	174.76	124.34
Radar Door Actuation	10.00	268.00	95.00
M.H. 64 Damping System	139.13	449.26	135.57
Radio & Radar Removable	1,912.00	180.31	103.97
Radio & Radar Fixed	631.00	233.27	115.71

continued

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Date: September 1st, 1957
Aircraft: C-105 MK 2
Production A/C

SECRET
UNCLASSIFIED
NON CLASSIFIED

Report # 7-0400-34 Iss. 11
Sheet # 3-2

WEIGHT & C.G. SUMMARY

DESCRIPTION	WEIGHT lb	H. ARM ins.	V. ARM ins.	M.A.C. %
Equipment (Fixed & Removable) (cont'd.)				
Sparrow Pack Structure	870.70	389.60	96.00	
Sparrow Pack Mechanisms	582.57	379.30	99.04	
Sparrow Pack Hydraulics	331.94	373.52	99.00	
Sparrow Pack Electronics	198.00	327.96	100.00	
Sparrow Pack Electrics	64.10	362.29	95.00	
Sparrow Pack Air Conditioning	2.64	309.93	104.15	
AIRCRAFT BASIC WEIGHT	42,239.24	546.36	128.40	
USEFUL LOAD (less fuel)	2,850.76	372.91	105.43	
Crew	430.00	194.00	136.50	
Oil	138.97	636.92	110.57	
Alcohol	22.00	93.00	138.00	
Engine Fire Extinguishing Fluid	25.00	730.00	129.00	
Residual Fuel	218.40	553.98	134.04	
Missiles	1,728.00	389.29	88.30	
Oxygen Charge	13.39	259.68	159.91	
Water for Air Conditioning	275.00	268.00	131.75	
Operational Weight Empty	U/C Up 45,090.00	535.40	126.95	27.46
	U/C Down	537.52	123.44	28.05
O.W.E. less Missiles	U/C Up 43,362.00	541.22	128.49	29.07
	U/C Down	543.43	124.84	29.68
Normal Combat Mission Fuel * (2,044 gals. @ 7.8 lb/gal.)	15,940.00	542.03	142.30	
	U/C Up	537.13	130.96	27.94
Normal Combat Weight	61,030.00	538.70	128.37	28.37
	U/C Down			
Half Combat Mission Fuel (1,022 gals. @ 7.8 lb/gal.)	7,970.00	543.18	139.55	
	U/C Up	536.57	128.84	27.78
Combat Weight (half Mission Fuel	53,060.00	538.37	125.86	28.28
	U/C Down			
Maximum Internal Fuel (2,492 gals. @ 7.8 lb/gal.)	19,438.00	541.85	144.16	
	U/C Up	537.34	132.13	28.00
A.U.W. Maximum Internal Fuel	64,528.00	538.82	129.68	28.41
	U/C Down			
Maximum External Fuel (500 gals. @ 7.8 lb/gal. + Drop Tank	4,248.00	521.84	60.67	
	U/C Up	536.38	127.72	27.73
A.U.W. Maximum Internal and External Fuel	68,776.00	537.77	125.42	28.12
	U/C Down			

N.B. 1) * New figures are being prepared by Aerodynamics for Mission Fuel in accordance with latest PS 13 data, but these are not yet available.

2) Aircraft Datum = 120" above and arbitrarily chosen ground line.

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WEIGHT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

PROJECT C-105 MK 2 Production A/C

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H. AF
TOTAL WING GROUP	1000000			
OUTER WING	1010000			
100 OUTER WING SKINS	1010001	64	92910706	
100 POSTS AND INTERCOSTALS	1010002	64	2436732	
100 OUTER WING SPARS	1010003	64	39258701	
100 OUTER WING RIBS	1010004	64	13456695	
18 AILERON CONTROL BOX	1010006	64	37920774	
100 RIB 12	1010007	64	7626648	
100 O W AILERON JACK FITTING	1010009	64	1046727	
AILERON JOINTS AND FAIRING	1010010		3293787	
			197945	
AILERON	1020000			
100 AILERON STRUCTURE	1020001	74	33600792	
			33600	
ELEVATOR MARRY UP	1030000		1597738	
97 ELEVATOR	1030000	82	44900755	
			46497	
INNER WING	1040000			
79 MAIN U C STRUCTURE	1040001	62	33071577	
27 FRONT SPAR I W	1040002	62	11110484	
27 MAIN SPAR I W	1040003	62	14639565	
27 FWD CENTRE SPAR I W	1040004	62	4245604	
27 AFT CENTRE SPAR I W	1040005	62	3386654	
27 REAR SPAR	1040006	62	7667703	
32 I W STRUCTURE F S TO M S	1040007	62	105918499	
I W STRUCTURE M S TO R S	1040008	62	14979609	
STRUCTURE AFT OF REAR SPAR	1040009	62	31639720	
11 ELEVATOR CONTROL BOX I W	1040010	62	42717725	
25 TRANSP JOINT MARRY UP	1040011		7065636	
74 CENTRE LINE JOINTS	1040012	62	58643661	
INNER WING SKINS	1040013	62	24204624	
STRUT PICK UP FITTINGS	1040014	62	11818615	
31 RIBS 1 TO 9 INNER WING	1040016	62	51204612	
DORSAL FAIRING	1040017	62	95482	
100 RIB 10	1040018	62	10800636	
FUSELAGE HINGE JOINT I W	1040019	62	13812621	
25 TRANSP JOINT	1040020	62	12942630	
FUSELAGE HINGE JOINT	1040021		3740527	
DORSAL FAIRINGS I W	1040023		5629572	
			669323	

WEIGHT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

REPORT NO. 7-0400-34 Iss. 11

Sept. 1/57

SHEET 4-1

	REFERENCE NO.	COMP. NO.	WEIGHT	H. ARM	V. ARM	H. MOMENT	V. MOMENT
	1000000						
	1010000						
	1010001	64	92910	70646	13597	65637199	12632973
LS	1010002	64	2436	73207	13410	1783323	326668
	1010003	64	39258	70118	13636	27526924	5353221
	1010004	64	13456	69506	13650	9352727	1836744
	1010006	64	37920	77409	13508	29353493	5122234
	1010007	64	7626	64873	13832	4947215	1054828
TING	1010009	64	1046	72791	13670	761394	142988
AIRING	1010010		3293	78726	13224	2592447	435466
			197945			141954722	26905122
	1020000						
	1020001	74	33600	79213	13556	26615568	4554816
			33600			26615568	4554816
	1030000		15977	73810	14260	1178746	227732
	1030000	82	44900	75523	14260	33909827	6402740
			46497			35088573	6630472
	1040000						
	1040001	62	33071	57757	13983	19100817	4624318
	1040002	62	11110	48428	14260	5380351	1584286
	1040003	62	14639	56570	14280	8281282	2090449
	1040004	62	4245	60417	14317	2564702	607757
	1040005	62	3386	65494	14446	2217627	489142
	1040006	62	7667	70374	14420	5395575	1105581
M S	1040007	62	1059	18499	2914405	52883798	15257488
R S	1040008	62	14979	60964	14572	9131798	2182740
R SPAR	1040009	62	31639	72041	14887	22793052	4710098
W	1040010	62	42717	72500	14327	30969825	6120065
P	1040011		7065	63621	13876	4494824	980339
	1040012	62	58643	66143	15062	38788239	8832809
	1040013	62	2242	0462440	14472	139992978	32446803
CS	1040014	62	11818	61500	14577	7268070	1722710
NG	1040016	62	5120	461287	14499	31381395	7424068
	1040017	62	9548	2880	16150	45866	15343
	1040018	62	10800	63683	13835	6877764	1494180
I W	1040019	62	13812	62104	13790	8577804	1904675
	1040020	62	12942	63071	13710	8162649	1774348
	1040021		3740	52796	14309	1974570	535157
	1040023		5629	57249	16026	3222546	902104
			669323			409505532	96804460

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REPORT NO. 7-0400-34 Iss. 11

SHEET 4-2

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WEIGHT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

PROJECT _____

[illegible]

AVRO AIRCRAFT LTD.
MALTON, ONT.

WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

PROJECT _____

DESCRIPTION		REFERENCE NO.	COMP. NO.	WEIGHT	H.
TOTAL FUSELAGE GROUP		3000000			
FUSELAGE BASIC STRUCTURE		3010000			
RADOME		3010001	51	17815	3
RADAR NOSE STRUCTURE		3010003	51	17444	9
SKIN F F		3010004	52	8094	19
7000	SKIN C F	3010005	54	22043	36
SKIN D B		3010006	56	5880	53
SKIN E B		3010007	58	16489	66
4	FORMERS F F	3010011	52	10531	18
FORMERS C F		3010012	54	28749	37
6	FORMERS D B	3010013	56	7331	53
LIGHT FORMERS E B		3010014	58	10370	56
8	HEAVY FORMERS E B	3010015	58	19214	66
LOWER LONGERONS F F		3010016	52	6846	18
INNER LONGERON E B		3010017	58	7184	65
LONGERON E B		3010018	58	6905	65
TOP LONGERONS E B		3010019	58	3866	66
LOWER SHEAR PANEL		3010020	52	2988	23
TOP WING LONGERONS D B		3010021	56	4985	34
LONGERONS C F		3010022	52	5325	57
LONGERONS C F		3010023	54	1381	34
LONGERONS D B		3010025	56	1056	95
ELECTRONICS BAY STRUCTURE		3010026	54	4523	27
59	LONGITUDINAL BEAMS D B	3010027	56	6570	52
RADAR ACCESS DOOR C F		3010028	54	2108	27
72	MAIN FRAME AT STN 538 77	3010029	56	5980	53
DIVE BRAKE ACCOM STRUCTURE		3010030	56	3634	50
20	TOP LONGERONS F F	3010031	52	1857	19
DUCT C F		3010032	54	3155	23
FLOATING DUCT D B		3010033	56	4160	05
CANOPY ARCHES		3010034	52	8535	22
4	ACCESS PANEL STIFFNERS	3010036	56	3865	53
66	NOSE U C SUPPORT STRUCTURE	3010037	52	3343	21
3	CREW BULKHEADS	3010038	52	1227	20
EQUIPMENT BAY STRUCTURE		3010039	54	7111	28
86	WINDSCREEN	3010041	52	1300	31
SERV ACCESS DOORS AND PNLS		3010042	58	5929	66
DORSAL FAIRING FWD		3010044	54	2727	29
INTERMEDIATE FORMERS E B		3010045	58	4873	72
ENGINE ACCESS DOORS E B		3010046	58	1073	56
SEALING MISSILE BAY		3010047	54	973	22
MISCELLANEOUS ITEMS C F		3010048	54	392	37
ARMAMENT BAY ROOF		3010049	54	1252	23
BULKHEAD AT STA 120		3010050	52	3060	12
BULKHEAD AT STA 255		3010051	52	1042	29
BULKHEAD STA 485		3010052	54	1235	14
ENGINE TUNNEL E B		3010054	58	6131	86
FUSELAGE FUEL TANK		3010055	54	2096	03
COCKPIT FLOOR		3010056	52	5743	17
DORSAL OVER FUEL TANK		3010061	54	5097	41

AVRO AIRCRAFT LTD.

MALTON, ONT.

WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

PROJECT _____

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H. A.
DEFLECTOR SHIELD	3010062	54	2000	337
SHEAR PANEL BELOW NAV ARCH	3010063	52	2074	251
PACK MOUNTING STRUCTURE	3010064	54	2365	292
ENG ACCESS DOORS FIXED R F	3010065	59	9326	772
TUNNEL FIXED R F	3010066	59	12072	771
SKIN AND INTERCOSTAL R F	3010067	59	5256	772
FAIRING UNDER RUDDER R F	3010068	59	2638	781
FORMERS FIXED R F	3010069	59	10681	780
LONGERONS C BEAM FIXED R F	3010070	59	2690	771
REMOVABLE NACELLES R F	3010071	59	3481	5822
CNTR STRUC STING REMOV R F	3010072	59	10794	834
MISC ITEMS STRUCTURE F B	3010073	58	9126	664
MTG BRKTS SIDE PANEL D B	3010075	56	2135	524
FORMERS LOWER PANEL D B	3010076	56	7416	544
LOWER PANEL D B	3010077	56	5054	554
JOINT F F TO C F	3010078		8552	55
JOINT AIR INTAKES TO F F	3010079		9522	220
JOINT C F TO D B	3010080		5004	85
JOINT D B TO E B	3010081		2508	584
FUS JOINT AT STA 742 50	3010082		4187	42
DROP TANK MTG STRUCT D B	3010083	56	8536	74
STRUCT AT ENG ACCESS DOORS	3010085	58	6537	640
MISC STRUCTURE F F	3010086	52	5521	87
ENG RAILS ATTACHM TO FORM	3010087	58	3624	685
			6580	59
96 DIVE BRAKES	3020000	72	11150	508
96 DIVE BRAKES C A	3020001		4064	87
			1155	6
AIR INTAKES	3030000			
AIR INTAKES	3030100	55	7550	2210
			7550	2
PILOTS AND NAVIG CANOPIES	3040000			
76 PILOTS CANOPY	3040100	53	24081	1176
			24081	
NAVIGATORS CANOPY	3040200	53	15650	232
			15650	

PROJECT_____

WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

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WEIGHT AND C. OF G BY FUNCTIONAL G
COMPONEN

PROJECT _____

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WEIGHT AND C. OF G BY FUNCTIONAL G
COMPONEN

PROJECT _____

DESCRIPTION		REFERENCE NO.	COMP. NO.	WEIGHT	H. /
FLYING CONTROLS GROUP		6000000			
MECHANICAL CONTROLS		6010000			
MECHANICAL CONTROLS F F		6010001	52	8759	158
MECHANICAL CONTROLS C F		6010002	54	8853	33
MECHANICAL CONTROLS D B		6010003	56	3504	86
MECHANICAL CONTROLS FIN		6010005	83	12688	787
11	MECHANICAL CONTROLS I W	6010006	62	40884	717
MECHANICAL CONTROLS O W		6010007	64	30436	770
MECHANICAL CONTROLS C A		6010008		646	709
				946	48
FLYING CONTROLS HYDRAULICS		6020000			
FLYING CONT HYDRAULICS D B		6020001	56	2332	574
37	FLYING CONT HYDRAULICS E B	6020002	58	39076	647
FLYING CONT HYDRAULICS R F		6020003	59	4875	58
63	FLYING CONT HYDRAULICS I W	6020004	62	24448	703
HYDRAULICS FLYING CONT O W		6020005	64	12181	740
61	FLYING CONT HYDRAULICS FIN	6020006	83	5331	761
				834	16

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MALTON, ONT.

WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

PROJECT _____

DESCRIPTION		REFERENCE NO.	COMP. NO.	WEIGHT	H.
BRAKE PARACHUTE I W		708000662		12462	
				6238	
ELECTRICAL SYSTEM		7090000			
ELECTRICAL WIRING		7091400			
ELECTRICAL WIRING R N		709140151		129	9
ELECTRICAL WIRING F F		709140252		2966	19
ELECTRICAL WIRING C F		709140354		2354	39
ELECTRICAL WIRING A I		709140455		1274	21
ELECTRICAL WIRING D B		709140556		9015	52
ELECTRICAL WIRING E B		709140658		6459	
ELECTRICAL WIRING I W		709140862		8909	54
ELECTRICAL WIRING FIN		709140983		3527	9
ELECTRICAL WIRING N U C		709141091		3301	8
60	ELECTRICAL WIRING ENG	7091411		2002	60
	ELECTRICAL WIRING C A	7091412		13746	33
				41141	
30	ELECTRICS F F	709150052		15248	15
				15248	
ELECTRICS C F		709160054		8709	45
				8709	
ELECTRICS E B		709170058		6706	7
ELECTRICS ENG		7091701		1753	25
ELECTRICS C A		7091702		7186	1
				19020	
ELECTRICS FIN		709180083		8389	
				83	
ELECTRICS I W		709190062		2336	62
ELECTRICS C A		7091901		1126	2
ELECTRICS M U C		709190292		7451	6
				2522	
ELECTRICS A I		709200055		5090	22
				5090	
ELECTRICS D B		709210056		30089	51
				30089	

IGHT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

REPORT NO. 7-0100-34 Iss. 11

SHEET 4-11

REFERENCE NO.	COMP. NO.	WEIGHT	H. ARM	V. ARM	H. MOMENT	V. MOMENT
708000662		124620222	14500		76907	17980
		6238			4799560	893524
7090000						
7091400						
709140151		129 9285	12000		11978	15480
709140252		296619977	11200		592518	332192
709140354		235439703	11000		934609	258940
709140455		127421984	12000		280076	152880
709140556		901552769	9921		4757125	894378
709140658		6459500	9200		38080	5888
709140862		890954111	13900		4820749	1238351
709140983		35279754	30000		280734	105600
709141091		33018600	9800		61380	32340
7091411		200260683	10000		1214874	200200
7091412		1374633473	11234		4601199	1544226
		41141			17593322	4780475
709150052		1524815464	11095		2357951	1691766
		15248			2357951	1691766
709160054		870945415	10760		3955192	937088
		8709			3955192	937088
709170058		67067944	11597		455225	77700
7091701		1763258919	11516		10388598	2036501
7091702		71861350	12000		440493	86160
		19020			11284316	2194361
709180083		8389550	28840		74327	23937
		83			74327	23937
709190062		233662053	14500		1449558	338720
7091901		11262946	13561		70500	15188
709190292		7451600	13700		38184	10138
		2522			1558242	364046
709200055		509022059	13832		1122803	704049
		5090			1122803	704049
709210056		3008951832	10258		15595730	3086530
		30089			15595730	3086530

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WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

PROJECT _____

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H.
WINDSCREEN DEMISTING	709220052		3564	19
			3564	
ELECTRICS R F	709230059		1368	0
			136	
SPARROW ELECTRICS F F	709240152		2541	8
SPARROW ELECTRICS C F	709240254		6027	
			314	
L P PNEUMATICS	7100000			
12 L P PNEUMATICS F F	7100000152		2117	19
12 L P PNEUMATICS C F	7100000254		7783	2
12 L P PNEUMATICS D B	7100000356		4185	5
12 L P PNEUMATICS E B	7100000458		1557	6
12 L P PNEUMATICS F IN	7100000583		5047	4
12 L P PNEUMATICS R N	7100000651		136	6
12 L P PNEUMATICS A P	7100000794		6029	
			5570	
INTAKE DEICING	712000055		5184	19
			5184	
RADOME ANTI ICING	713000051		888	5
			888	
RADIO AND RADAR FIXED	7140000			
RADIO AND RADAR FIXED R N	7140000151		16500	9
RADIO AND RADAR FIXED F F	7140000252		5000	18
RADIO AND RADAR C F	7140000354		19000	26
RADIO AND RADAR FIXED D B	7140000456		1800	51
RADIO AND RADAR FIXED F IN	7140000583		3200	76
RADIO AND RADAR FIXED C A	71400006		17600	21
			63100	
29 CANOPY ACTUATION	716000052		6492	22
			6492	
CABIN CONSOLES	717000052		1745	17
			1745	

MALTON, ONT.

COMPONE

PROJECT _____

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H.
RADAR DOOR ACTUATION	718000054		1000	26
			1000	
SURFACE FINISH F F	7190000152		363	18
SURFACE FINISH CANOPY	7190000253		6420	
SURFACE FINISH C F	7190000354		1400	37
SURFACE FINISH 2NTAKES	7190000455		40822	
SURFACE FINISH D B	7190000556		37753	
SURFACE FINISH E B	7190000658		44366	
SURFACE FINISH R F	7190000759		64880	
SURFACE FINISH I W	7190000862		315160	
SURFACE FINISH O W	7190000964		106770	
SURFACE FINISH O W L EDGE	7190001065		31963	
SURFACE FINISH AILERON	7190001174		35379	
SURFACE FINISH ELEVATOR	7190001282		56675	
SURFACE FINISH FIN	7190001383		3974	
SURFACE FINISH RUDDER	7190001484		20281	
			10000	

EIGHT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

REPORT NO. 7-0400-34 Iss. 11

SHEET 4-13

	REFERENCE NO.					COMP. NO.	WEIGHT		H. ARM		V. ARM		H. MOMENT		V. MOMENT				
	7	1	8	0	0	0	5	4	10	00	268	00	95	00	268	00	95	00	
									10	00					268	00	95	00	
	7	1	9	0	0	0	1	5	2	3	6	3	187	44	122	00	680	41	
	7	1	9	0	0	0	2	5	3	6	4	2	02	18	159	00	129	40	
	7	1	9	0	0	0	3	5	4	14	00	371	97	124	55	520	75		
S	7	1	9	0	0	0	4	5	5	40	8	22	79	81	20	20	930	16	
	7	1	9	0	0	0	5	5	6	3	7	7	53	82	81	12	00	20	29
	7	1	9	0	0	0	6	5	8	4	4	3	66	70	11	20	00	29	54
	7	1	9	0	0	0	7	5	9	6	4	8	80	36	21	31	75	52	07
	7	1	9	0	0	0	8	6	2	3	1	5	16	01	44	14	40	00	18
	7	1	9	0	0	0	9	6	4	10	6	7	70	65	01	35	90	75	38
EDGE	7	1	9	0	0	1	0	6	5	3	1	9	63	94	81	35	26	20	39
N	7	1	9	0	0	1	1	7	4	3	5	3	79	19	01	35	56	27	95
OR	7	1	9	0	0	1	2	8	2	5	6	6	75	52	31	42	60	42	74
	7	1	9	0	0	1	3	8	3	3	9	7	46	63	20	83	7	47	70
	7	1	9	0	0	1	4	8	4	2	0	2	81	27	92	14	31	16	41
									100	00					59	15	16	7	

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WEIGHT AND C. OF G BY FUNCTIONAL COMPONENTS

PROJECT _____

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H.
REMOVABLE EQUIPMENT GROUP	8000000			
EJECTOR SEATS	8010100	52	2844220	
			28442	
RADAR REMOV GROUP	8020000			
RADAR REMOVABLE	8020100			
RADIO AND RADAR REM R N	8020101	51	102070	9
RADIO AND RADAR REM F F	8020102	52	17860	18
RADIO AND RADAR REM C F	8020103	54	61520	27
RADIO AND RADAR REM D B	8020104	56	5750	51
RADIO AND RADAR REM C A	8020106		4000	55
			191200	
DAMPING SYSTEM	8030000			
DAMPING SYSTEM F F	8030101	52	635	16
DAMPING SYSTEM C F	8030102	54	2180	31
DAMPING SYSTEM E B	8030103	58	369	63
DAMPING SYSTEM I W	8030104	62	5800	52
DAMPING SYSTEM O W	8030105	64	957	4
DAMPING SYSTEM R N	8030106	51	520	8
DAMPING SYSTEM C A	8030107		4314	48
			13913	
SPARROW PACK	8050000			
SPARROW PACK STRUCTURE	8050100	94	87070	38
			87070	
SPARROW PACK MECHANISMS	8050200	94	58257	37
			58257	
SPARROW PACK HYDRAULICS	8050300	94	33194	37
			33194	

WEIGHT AND C. OF G BY FUNCTIONAL COMPONENT

PROJECT.

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WEIGHT AND C. OF G BY FUNCTIONAL
COMPONE

PROJECT _____

DESCRIPTION	REFERENCE NO.	COMP. NO.	WEIGHT	H
OPERATIONAL LOAD	9000000			
CREW MEMBERS	901000052		4300019	
			43000	
ENGINE OIL	9020000			
ENGINE OIL DRAINABLE	9020001		929065	
ENGINE OIL TRAPPED ENG	9020002		177363	
ENGINE OIL TRAPPED D B	902000356		192558	
ENGINE OIL TRAPPED E B	902000458		90959	
			13897	
SPARROW MISSILES	903000054		17280038	
			172800	
OXYGEN CHARGE	9040000			
OXYGEN CHARGE F F	904000152		7220	
OXYGEN CHARGE C F	904000254		126720	
			1339	
ALCOHOL FOR RADOME DEICING	905000051		2200 9	
			2200	
ENGINE FIRE EXT FLUID	9060000		250071	
			2500	
RESIDUAL FUEL	9070000			
RESIDUAL FUEL C F	907000154		468039	
RESIDUAL FUEL I W	907000262		1716059	
			21840	
WATER FOR AIR CONDITIONING	910000054		2600020	
UNUSABLE WATER FOR AIRCOND	910000154		150020	
			27500	

WGT AND C. OF G BY FUNCTIONAL GROUP COMPONENT

REPORT NO. 7-0100-34 Iss. 11

SHEET 4-16

	REFERENCE NO.	COMP. NO.	WEIGHT	H. ARM	V. ARM	H. MOMENT	V. MOMENT
	9000000						
	901000052		43000	19400	13650	8342000	5869500
			43000			8342000	5869500
	9020000						
	9020001		9290	65225	11390	6059403	1058131
	9020002		1773	63676	11640	1128975	206377
	902000356		1925	58084	9555	1118117	183934
	902000458		909	59928	9704	544746	88209
			13897			8851241	1536651
	903000054		172800	38929	8830	67269312	15258240
			172800			67269312	15258240
	9040000						
	904000152		722	20493	13200	14755	9504
	904000254		1267	26280	16150	332968	204621
			1339			347723	214125
CING	905000051		2200	9300	13800	204600	303600
			2200			204600	303600
	9060000		2500	73000	12900	1825000	322500
			2500			1825000	322500
	9070000						
	907000154		4680	39424	12300	1845043	575640
	907000262		17160	59755	13705	10253958	2351778
			21840			12099001	2927418
NING	910000054		26000	26800	13200	6968000	3432000
COND	910000154		1500	26700	12700	400500	190500
			27500			7368500	3622500

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