

AVRO AIRCRAFT LIMITED INTER-DEPARTMENTAL MEMORANDUM

Ref: 9354/11/J

Date: 1 June, 1958

To: See Distribution

From: E.F. Burnett - Weight Supervisor

Subject: ARROW 2 PRODUCTION A/C WEIGHT & C.G. SUMMARY - Report 7-0400-34 Issue 20

Attached is a copy of Weight and C.G. Summary Report 7-0400-34 Issue 20 dated June 1st, 1958, for your retention.

This report is revised monthly and is issued complete on the 1st of each month.

Classification cancelled / changed to: UNCLASSSIFIED

By authority of: DRDA 7/DARFT 5-8/DAS Eng 6-4-5

Date: 5 Nov 1992

Signature: Bauliney

Unit / Rank / Appointment: DSIS 3, Secretary CRAD HQ DRP

EFB/ag

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Aeronautical

UNCLASSIFIED RET CLASSIFIE

Aircraft: ARROW 2 Prod. A/C Date: 1 June, 1958

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1-1 to 1-5	Introductory notes and explanation of weight changes
2-1 to 2-2	Weight and C.G. Summary
3	Horizontal C.G. plot showing basic fixed points on flight envelope. The possible variations of C.G. with fuel used has been omitted until such time as fully approved fuel sequencing is established.
4-1 to 4-15	I.B.M. Detail sheet of Weight and C.G.s.

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## INTRODUCTION & WEIGHT CHANGES

The following is a Weight and C.G. Summary of the Arrow 2 Production Aircraft based on the latest weight estimates available.

All weight and C.G. changes are relative to Issue 19 of May 1st, 1958.

Some early Arrow 2 Aircraft, serial numbers in the group 25206 to 25236, will not be as this summary designates, but will be Flight Test Aircraft with Instrumentation, "Astra Minus", or preproduction Astra I Radar, Missile pack trials and various test installations etc. Appropriate weight statements will be issued at a later date.

#### GENERAL: -

- a) Orenda PS13 Engines comprise the Power Plant (4,500 lb each, excluding Nose Bullet & Input Frame).
- b) A package containing 4 "semi-submerged" Sparrow II missiles (432 lb each) forms the current Armament.
- c) The R.C.A. Astra I Radar system is installed. The basis of the unit weight & C.G. breakdown is a weight statement received from R.C.A. dated October 4, 1957. No Sparrow III auxiliaries are carried in a production Aircraft. Where later weights have been received from Minneapolis-Honeywell, these have been incorporated.

All installation cables, junction boxes etc not R.C.A. supplied are now AVRO's responsibility weightwise and these have been recently reassessed.

A weight statement has been received from R.C.A. dated April 1, 1958, but this will not be used until several points have been clarified by R.C.A. Some weight increase is to be expected.

The Infra Red tracker system seeker head is installed in its fin pod location.

(Total weight of the installed system is 3,349.4 lb, including missile actuation and firing systems).

In addition to the above Minneapolis-Honeywell MH64 Damping System is installed.

Arrow 2 A/C 25206 to 25208, and some later serial numbers will have the "Astra Minus" system (Navigational and Communication equipment only).

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## GENERAL: - (Cont d)

d) Where actual weights of Arrow I parts that apply to Arrow 2 Aircraft have been obtained, these weights have been recorded in the Arrow 2 records. Some weights of Arrow 2 parts have also been obtained.

On I.B.M. sheets in this report, immediately preceeding the item title will be found a number varying from 0 to 100. This is precentage actual weight recorded in the relevant item.

e) To increase the long range capabilities of the Aircraft a tailcone plug has been introduced to restrict the exhaust area. This plug will be used for ferrying missions only and its weight is recorded with the Max. Gross Weight see sheet 2-2.

#### STRUCTURE: -1.

WEIGHT 1b

### Wings

I/W Centre Line Joints - Mk 2 prod. drg. est of joints Main Spar to Auxiliary Spar	+	0.14
Elevators - Mk. 2 prod. drgs est, some redesign compared		9 .50
to Mk 1 - no allowance yet made for split		
control surfaces until design is finalized	=	6.98
Wing Weight Decrease	_	6.84

### b) Fin & Rudder

No weight change.

# c) Front Fuselage Fwd Sta. 255 ins

Navigator's Canopy* - Sierracin 880 replaces glass for		
the window	_	6.26
Windscreen* - addition of rear fairing apron	+	1.55
addition of shims & alterations to		
hardware for installation	+	0.51
Air Intakes - Subsequent to supersonic flight the .04		
Mg outer side skins were found to be cracke	d.	
Repair scheme instituted, skins now		
.04 AL, stringer sections changed		
backing angles added etc.	+	6.42
	CHESCAGO	
Front Fuselage Increase	+	2.22
CONTRACTOR OF THE PROPERTY OF	0.00	

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## STRUCTURE (Cont d)

WEIGHT 1b

N.B. \*The use of Sierracin 880 and Sierracete in place of glass and thermapane is being investigated for the windscreen and pilots canopy. If it proves acceptable and NO change in overall thickness is made then there would be weight savings of approx.

a) 32 lb to Pilots Canopy

) Total 70 lb.

b) 38 lb to Windscreen

(Sierracin = .045 lb/cu.in.

Glass = .090 lb/cu.in.)

d) Centre Fuselage Sta. 255 -485 ins

Duct CF - due to a D.O. error the splice straps at Sta. 257.45 were originally shown as .032 instead of .050 AL.

+ 0.73

Centre Fuselage Increase

+ 0.73

e) Duct Bay Sta 485 - 591.65 ins

No weight change.

f) Engine Bay Sta. 591.65 - 742.5 ins

No weight change.

g) Rear Fuselage Sta. 742.5 ins aft

No weight change.

h) Fuselage "Marry-Up"

Joint Duct Bay to Eng. Bay - Mk. I weight was recorded here until Mk 2 drgs were available. The Mk. I joint drgs. included some of the heat exchanger area, not applicable to Mk 2 A/C - the weight was left in error. Mk 2 drgs now

estimated

- 10.87

Fuselage "Marry-Up"

-10.87

TOTAL STRUCTURAL DECREASE

- 14.76

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2.	LANDING	GEAR

WEIGHT 1b

Main Landing Gear - increase made to brake weight allowance assuming redesigned Mk 2 brakes to be simular to those for MK 1 A/C. There are probably further increases to wheels and brakes this is only a preliminary allowance until a definite design proposal is accepted. (Wt of ONE wheel, brake & tyre Assy. was 151.46 lb, now 165 lb allowed)

+ 54.16

TOTAL LANDING GEAR INCREASE

54.16

#### 3. POWER PLANT & SERVICES

Engine Controls - completely re-estimated to Mk. 2 prod. drgs.

Fuel System - completely re-estimated to Mk 2 drgs previously allowance based on Mk 1 and scheme drgs.

0.81

0.11

POWER PLANT & SERVICES DECREASE

0.70

## FLYING CONTROLS GROUP

No weight change.

## 5. EQUIPMENT GROUP

Air Conditioning - first complete estimate made to Mk 2 prod. drgs, previously allowances etc based on

Mk 1 system were carried. No attempt will be made to analyse the weight change which

is relatively small 8.09 Electrics Duct Bay - Further details available on E28 panel

- now fully estimated to prod. drgs. Radio & Radar Fixed - deletion of 4 Radar Homer antennae complete re-estimate of cables, the

weight prediction of which is now AVRO

responsibility (see introd. note c) + 67.45 Missile Pack Structure - addition of retraction guides misc

E.C.N.'s incorporated + 6.49 Missile Pack Hydraulics - pressure return lines now Steel

were Aluminum, also CS-T-162 pipe replaces CS-T-135 throughout system

5.46

4.39

# TOTAL EQUIPMENT GROUP INCREASE

+ 83.40

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## SUMMARY

## Weight Change - Aircraft Basic Weight

- 14.76 Structure + 54.16 Landing Gear Power Plant - 0.70 Equipment + 83.40

+123.50

# Weight Change - Operational Weight Empty (A/C Less Fuel)

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46,229.54 46,353.04

+123.50 lb

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Aircraft:

ARROW 2

Date:

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DESCRIPTION	WEIGHT	H. ARM	V. ARM
	lb	ins	ins
STRUCTURE	19,154,35	566.68	137.31
Wings	10,029.89	642.98	142.21
Fin & Rudder	1,039.01	735.84	208.90
Fuselage - Fwd Sta 255 ins	2,588.02	184.43	128.67
Sta 255 - 485 ins	1,700.93	376.01	130,59
Sta 485 - 591.65 ins	1,161.25	538.59	105.63
Sta 591.65 - 742.5 ins	1,574.07	659.68	110.72
Sta 742.5 ins aft	1,019.48	806.57	127.46
"Marry - Up"	41.70	460.70	110.60
LANDING GEAR - RETRACTED	2,638.41	488.35	134.95
Main Landing Gear	1,989.10	538.89	141.00
Main Gear Doors & Fairings	287.98	538.52	138.40
Nose Landing Gear	333.81	170.81	99.70
Nose Gear Door & Fairing	27.52	162.22	88.66
POWER PLANT & SERVICES	10,801.37	671.45	121.35
Engine & Accessories PS13	9,186.78	687.95	121.16
Engine Nose Bullets (Orenda Supplied)	70.00	587.17	116.00
Engine Controls	32.32	379.90	117.49
Gear Box & Drives on Fuselage	281.84	601.70	102.49
Gear Box, Starter & Drives on Engines	315.45	615.98	105.24
Fire Extinguishing System	65.46	700.45	134.21
Engine Mountings	132.38	666.82	136.52
Fuel System		531.46	
FLYING CONTROLS GROUP	717.14 1,926.56	650.93	134.96 138.14
Mechanical Flying Controls	952.74	676.42	147.63
Hydraulic Flying Controls	973.82	626.00	128.86
EQUIPMENT FIXED & REMOVABLE	9,033.59	331.29	114.21
Instruments	46.07	163.68	138.70
Probe	15.25	-23.71	108.00
Cockpit Pressure Sealing	5.00	186.00	130.00
Oxygen System	26.07	240.54	156.70
Cockpit Equipment	6.00	187.50	145.00
Ejector Seats	342.94	204.50	134.11
Air Conditioning System	864.09	341.68	135.33
Cockpit Insulation	14.31	187.48	132.00
Hydraulics Utilities system	647.33	505.17	117.69
Drag Chute	91.07	786.68	143.19
Electrical System	1,268.41	434.75	112.77
Low Pressure Pneumatics	56.94	421.96	128.47
Surface Finish	100.00	591.52	140.20
Intake Deicing Boots	88.00	195.82	118.00
Canopy Actuation	64.92	221.99	154.35
Cabin Consoles	17.28	174.66	124.33
Radar Door Actuation	10.00	268.00	95.00
MH64 Damping System	184.72	464.22	135.05
Radio & Radar Removable	2,095.20	and the second s	
Radio & Radar Fixed		209.73	108.59
Sparrow Pack Structure	841.53	212.59	114.94
Sparrow Fack Mechanisms	897.36 553.32	383.98 372.85	98.77 99.13
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Aircraft:

ARROW 2

Prod. A/C

Dates

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DESCRIPTION	WEIGHT	H. ARM	V. ARM	% M.A.C.
22001412401	lb	ins	ins	/S
Sparrow Pack Hydraulics	365.51	378,03	103.89	
Sparrow Pack Electronics Remov.	185.50	331.91	100.00	
Sparrow Pack Electrics & Electronics	217.11	335.24	110.00	1
Fixed				i l
Sparrow Pack Air Conditioning	5.21	305.50	102.50	
Sparrow Pack L.P. Pneumatics	24.45	386.92	103.50	
U/C Up		542.82	128.45	
Aircraft Basic Weight	43,554.28			
U/C Down	,	545.13	124.64	
USEFUL LOAD (less fuel)	2,798.76	377.29	104.80	
Crew	390.00	194.00	136.50	
	138.97	636.92	110.57	
Engine Fire Ext. Fluid Residual Fuel	25.00	730.00	129.00	
Missiles	218.40 1,728.00	553.98 389.29	134.04 88.30	
Oxygen Charge	13.39	259.68	159.91	
Water for Air Conditioning	285.00	267.91	131.56	
U/C Up	20).00	532.83	127.02	26.75
Operational Weight Empty	46,353.04	772.07	121002	2001)
U/C Down	4-9000004	535.00	123.44	27.35
U/C Up		538.39	128.52	28.29
Op. Wt. Empty less Missiles	44,625.04			
U/C Down		540.64	124.80	28.91
Normal Combat Mission Fuel*	17,580.00	60	63	
(200 N. Mile - $2,254$ gal at $7.81$ gl				
Normal Gross Weight	63.933.04		-	
Half Combat Mission Fuel	8,790.00	-	-	
(1,127 gals at 7.8 lb/gal)	4			
Combat weight (half mission fuel	55,123.04			
Max. Internal fued	19,438,00	541.85	141.16	4. *·
(2492 gals at 7.8 lb/gal)		22212	122 20	00 10
U/C Up	65 707 OL	535.49	132.08	27.49
Gross Weight (Mak. int. Fuel) U/C Down	65,791.04	537.02	129.56	27.91
Max External Fuel (500 gal	4,242.36	520.32	60.79	<1.07⊥
at 7.81b/gal + Drop Tank)	49242000	720.72	00.17	
Tailcone Plugs	252.00	857.07	128.60	
		535.73		27.55
II/G IIn	1			
U/C Up Max. Gross Weight (Int & Ext fuel)	70,285.40	222012	127.76	41.000

N.B. 1) Aircraft Datum = 120 ins above an arbitrarily chosen ground line

<sup>2) \*</sup> Fuel weights in accordance with latest data issued by Performance Group dated November 19th, 1957. Centres of gravity have temporarily been omitted until a fuel sequencing system is finally established.

