





A. V ROE CANADA LIMITED
MALTON ONTARIO

ANALYSED

TECHNICAL DEPARTMENT (Aircraft)

AIRCRAFT:

REPORT NO. 7-0558-63

FILE NO.

~~CONFIDENTIAL~~

NO. OF SHEETS:

12

Classification cancelled / Changed to UNCLASS

By authority of AVES

Date 30 Sept 96

Signature [Signature]

Unit / Rank / Appointment AVES

TITLE:

NATURAL FREQUENCIES UNDER LATERAL LOAD OF FUSELAGE FORMERS

Stn. 591.65, 644.43, 697.28, 742.50
485.00 538.77

NRC - CISTI
AÉRO / M.E.
LIBRARY

87- 12 10

BIBLIOTHÈQUE
AÉRO / G.M.
CNRC - ICIST

PREPARED BY [Signature] DATE 25/1/56

CHECKED BY DATE

SUPERVISED BY [Signature] DATE Feb. 56.

APPROVED BY DATE

| ISSUE NO | REVISION NO | REVISED BY | APPROVED BY | DATE | REMARKS |
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AVRO AIRCRAFT LIMITED
MALTON - ONTARIO

TECHNICAL DEPARTMENT

REPORT NO. 7-0558-63

SHEET NO. 1

AIRCRAFT:

C 105

PREPARED BY

DATE

CHECKED BY

DATE

Calculation of Natural Frequency **CONFIDENTIAL**

The Natural Frequency f , in cycles/sec., is given by

$$f = \frac{1}{2\pi n} \sqrt{\frac{k}{m}}$$

where: k = spring constant of structure, in lb. of load per inch of deflexion.

m = mass of structure and equipment, in lb.sec./in.

The values of k , for the five formers under consideration, were obtained in the following reports:

str. 591.65 Rep. 7-0558-47

644.43 7-0558-62

697.28 "

742.50 "

538.77 "

A. V. ROE CANADA LIMITED
MALTON - ONTARIO

TECHNICAL DEPARTMENT (Aircraft)

REPORT No. 7-0558-63

SHEET No. 3

AIRCRAFT:

C 105

STA 538.77 - 742.50

PREPARED BY

DATE

CHECKED BY

DATE

21/12/55

Weights of Items Mounted Below Φ Duct, Between above Sta.

| STA. | ITEM | WEIGHT to nearest lb. |
|---------------------|---------------------------------------------------------------------------------------|-----------------------------|
| 509.50 to 567.50 | Hydraulic Piping (wet) | 21 |
| 485 538 | } Dive Brake Jacks | 33 |
| 491 | 2 Service Panels i) Refuel + Test ii) Aft Circuit Limiter | 8 |
| 524 538 | } Main Alternator + Rectifier Panels (2 Panels + Boxes) | 175 |
| 514 519 | } } Ground Fault Relays (2 off) | 9 |
| 529 534 | } } | |
| 539 | 2 Small Values i) Dive Brakes Selector ii) Emergency by pass for Utility System | 6 |
| 542 | V/C Selector valve | 4 |
| 548 | Heat Exchanger (Fuel to Oil) Fuel Proportioner | 100 |

356.

A. V. ROE CANADA LIMITED
MALTON - ONTARIO

TECHNICAL DEPARTMENT (Aircraft)

REPORT NO. 7-0558-63

SHEET NO. 5

AIRCRAFT:

C105

STA 53877-74250

PREPARED BY

DATE

CHECKED BY

DATE

21 12 55

Weights of Items Mounted Below ϕ Duct, Between above Sta.

| STA | ITEM | WEIGHT to nearest lb |
|---------------------|---------------------------------------------------|----------------------------|
| 673.25 to 717.36 | Hydraulic Piping (wet) | 5 |
| 697 707 | Hydraulic Accumulators (Flying Controls) 2 off | 44 |
| | Control valves 2 off | 8 |
| | Pressure Switches 2 off | 3 |
| 690 | Filters 2 off mounted on beam at ϕ duct | 15 |

75

N.B. this table is based on information received from D.O.



AVRO AIRCRAFT LIMITED
MALTON, ONTARIO

TECHNICAL DEPARTMENT

AIRCRAFT:

C 105

FORMER STN
485.00

REPORT NO. 7-0558-63

SHEET NO. 6

PREPARED BY

Doherty

DATE

29/2/56

CHECKED BY

DATE

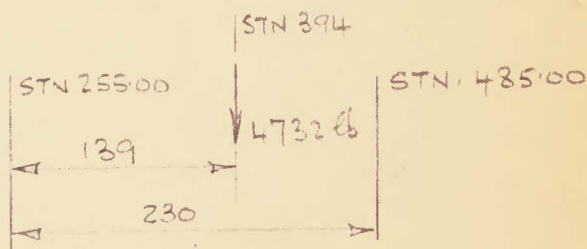
Estimated Weights of items between
stn. 370.00 and 509.50

Armament Pack: 3700 lb. (D. Turner 29/2/56)
assume that half of this (1850 lb) is applied
at stn. 485.00

Fuel Tank + Fuel:

Tank structure: 250 lb assume @ stn 394.31
cells and fuel: 4482 lb. @ stn. 394.31
(K. Griffin & J. Watson 29/2/56)

Total: 4732 lb @ stn 394.31



$$\text{Reaction @ stn. 485.00} = \frac{139 \times 4732}{230} = \underline{2860 \text{ lb}}$$

$$\text{Structure (below } \phi \text{ duct)} = \underline{696 \text{ lb}}$$

(J. Maclean 29/2/56)

$$\text{Power Panel} = 70 \text{ lb. @ stn. 493}$$

(A. Shaw 29/2/56)

