Date: December 1st, 1956 Aircraft: C-105 With J75

PS Engines as Interim Power Plant

UNCLASSIFIED

Report # 7-0400-05 Sheet # 002-1 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

NON CLASSIFIE
WEIGHT CHANGES TO 7-0400-05 ISSUE 26
of May 1st, 1956

WEIGHT CHANGES

Wing:

+	95.01	Structure for Main U/C - Completely re-estimated, previous allowances were unrealistic. Side Stay Attachment Aft Pivot Jack Pick-Up Main Pivot fitting Bushings & bearings main pivot Main U/C Up-locks - omitted	++++	2.16 8.74 28.32
•	15.22	I/W Leading Edge - Completely estimated to Production Drawings An allowance for doublers at O/B end was carried - these do not exist Hardware - previous allowance too low Cuff Assy. Stiffeners - Allowance too low Skins - Allowance too low	* + + + + -	
+	144.51	I/W Structure F/S to M/S - Estimated mainly to Production Drawings. Tanks # 3 & 4 early estimates omitted most of internal structure, trusses etc. 2 Point refuelling door added U/C Door Up-locks - allowance was too low Fixed Structure pivot door etc, included F/S Hardware 20 lb of which was deleted from F/Spar (see I/W Spars), largley omitted Joint U/C Fitting L/E & F/Spar added here Fuselage Side Rib - re-estimated Aux. Rib A/C C/L, Aux. Spar to Main Spar duplicated in C/L Joints. Aux. Spar & Joint to Fuselage side 10 Perestimate Joint Aux. Spar to Front Spar Joint Aux. Spar at A/C C/Line Other Misc. Changes	+ + + + + 0 0 0 + + + 0	8.50 0.24 1.11

- 2,581.21 I/W Structure M/S to R/Spar - Skins, ribs & Strut pick-ups are now in separate reports see below, however, remaining structure increased by + 59.64 lb due to underestimation of miscellaneous internal structure.

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Report # 7-0400-05 Sheet # 002-2 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

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WEIGHT CHANGES TO 7-0400-05 ISSUE 26 of May 1st, 1956

WEIGHT CHANGES

Wing:

Comme				
+ 2	2,179.80	I/W Skins - This is a new report, Skins were in M/S to R/S Structure. The skins have been completely re-estimated to nominal thicknesses, sinc checked by Actual Weighings. Machined skins - gauge changes Access Doors - accrease in gauge of pump doors etc. Hardware - previous allowances were far too low, now fully detailed	; ee +	10.47 7.66 94.12
		Actual Increase in Skins	+	96.93
,+	517.56	I/Wing Ribs # 1=9 - New report were included in M/S to R/Spar Structure Actual Weight Changes are as follows: Rib.# 1 - Included hardware allowance and some strut. pick-ups etc.	an a	13.14
		Rib.# 2 - redesigned machined fittings added		
		Rib.# 3 - reinforced, capping etc. in-	+	32.14
		creased for duct pick-up Rib.# 4 - redesigned Aft Web .125 was .156 fwd. web .09 was .12, delete 5.8 lb former pick-up, too heavy	+	22 • 44
		hardware etc.	æ	36.26
		Ribs. # 5 to 9 - Minor Changes	+	1.29
		Actual Changes	+	6.47
=	31.04	I/W Spars - Re-estimate of Main Spar	_ 	3.44
		Re-estimate of C/F, C/R, Front and Rear Spars	-	7.60
		F/Spar hardware now in F/S to M/S Structure.	E	20.00
			co	31.04
+	55.68	I/W Fuselage Pick-Up Brackets - New Report, were included in M/S to R/S Structure. Re-estimate of former pick-ups with		
		Rib.#4.	+	1.72
		Re-estimate of Engine Mounting Pick-Ups	+	4.76
		Marry-Up bolts struts to fittings omitted	+	2.82
		Miscellaneous changes	=	4.02
		Duct Pick-Up brackets - omitted Strut Pick-Up at 538.77 - omitted	+	11.96
			+	9.02
		Actual Changes	+	26.26

Date: December 1st, 1956 Aircraft: C-105 With J75 P5 Engines as

Interim Power Plant

Report # 7-0400-5 Sheet 002-3 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

ASSIFIE ISSUE 26 of May 1st, 1956

WEIGHT CHANGES

Wing:

	347.97	Joints I/W - Delete joint to Fuselage - now separate report see below Re-estimate of C/L Joint including thrust mounting which was in M/S to R/S Structure. Delete Trans. Joint allowance and	- 146.70 + 38.13
		Rib.# 10 - now considered separately below	- 239.40 - 347.97
+	110.44	Rib.# 10 - New report - was included in Wing Joints - see above, completely restinated - actual increase	+ 0.44
+	179.38	I/W Hinge Joint to Fuselage - New report was in Wing Joints - see above Stainless Steel Seal added at Fus. side Rib. Hardware allowance was too low Wing to C.F. Joint - omitted Joint Aft of R/Spar - omitted Miscellaneous changes Actual Change	+ 14.25 + 4.58 + 8.76 + 7.70 = 2.61 + 32.68
+	29.30	Elevator - An Actual Weight was obtained Recent addition of adjustment plates etc. Recent addition of links at attach. Omission of superseded design Marry-up	+ 9.33 + 4.52 + 4.00 + 11.45 + 29.30
+	4.84	I/W Elevator Control Box - Re-estimate some Actual Weights Joint of Centre Box was omitted	- 1.34 + 6.18 + 4.84
8	2.39	I/W Structure Aft R/Spar - entirely re-estimated, mainly to Production Drawings.	
+	57.64	I/W Dorsal Fairing - New report was included with Wing Fairings below. Complete re-estimate to Schemes, previous allowance was 45 lb., hence actual increase	+ 12.64

Continued.

Date: December 1st, 1956 Aircraft: C-105 With J75 P5 Engines as

UNCLASSIFIED

Report # 7-0400-05 Sheet # 002-4 Issue 27 Prepared By: L. Griffin Checked By: E. Burnett

WEIGHT CHANCES TC LOAD SOS Issue 26 of May 1st, 1956

WEIGHT CHANGES

Wing:

+	99.81	Transport Joint I/W to O/W - Previously titled Wing Fairings Aileron Link fairings now called up with Airleron Marry-up (see below) Delete Fairings at Dorsal area (see above) Complete re-estimate of Transport Joint Fairing at Joint - previous allowance too low	- 16.80 - 45.00 - 143.61 + 18.00 + 99.81
		* It should be noted here that 71.10 lb butt straps were allowed on Rib.# 12 0/Wing (now deleted - see below) also 121.0 lb on Wing Joints (see above) hence actual decrease of -48.49 lb on complete Transport Joint.	+ 99.81
0	69.54	O/W Rib.# 12 - Delete butt straps and hardware now included in Transport Joint (see above) Miscellaneous Changes to Production Drawings.	- 71.10 + 1.56 - 69.54
+	125.21	O/W Skins - Completely re-estimated to Production Drawings Addition of stringers (see below) Addition of Attach. to ribs (see also O/W Ribs.) Addition of Hoist fittings Correction to D.O. errors in hardware call up etc.	+ 104.52 + 13.90 + 2.32 + 4.47 + 125.21
œ	100.27	O/W Posts & Intercostals - this report used to contain stringers now called up with skins Removal of Stringers (see above) Re-estimate of Posts Re-estimate of Intercostals	- 97.85 - 1.15 - 1.27 - 100.27
	11.68	O/W Ribs - Drawings checked to latest Production Issues. Removal of Aileron Jack fitting (see below) Skin Attach. now with skins	+ 12.68 - 10.46 - 13.90 - 11.68

Date: December 1st, 1956 Aircraft: C-105 With J75

P5 Engines as UNCLASSIFIED Interim Power Plant NON CLASSIFIE
WEIGHT CHANGES TO 7-0400-05 Issue 26

Report # 7-0400-05 Sheet # 002-5 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

of May 1st, 1956

WEIGHT CHANGES

W	i	n	ø	•
	-	44	-	0

Win	ng:			
+	10.46	Aileron Jack Fitting - New report was previously included with Ribs (see above)		
+	34.56	O/W Leading Edge - Marry-Up to F/Spar omitted Estimated entirely to Production Drawings, previous estimate to	. +	4.59
		early schemes	+	29.97
			+	34.56
can	14.18	O/W Aileron Control Box - completely checked to Production Drawings, too high an allowance was previously made for stiffeners and hardware.		
+	5.24	O/W Spars - Spar joints to Rib # 12 omitted Production Drawing estimate of	+	3.20
		Main Spar Production Drawing estimate of	æ	6.82
		C/Spar fwd.	+	4.36
		Decrease in caps of F/Spar Production drawing estimate of	æ	2.89
		C/Spar aft. Rear Spar hardware allowance etc.	+	1.78
		too low	+	5.61
			+	5.24
+	5.96	Aileron - Completely checked, skins were underestimated,		
		many actual weights of parts etc.	+	14.92
		Marry-up bolts now included separately	œ	8.86
			+	5.96
+	32.93	Aileron Marry-Up - New report includes all bolts and fair- ings above allowances for which were too low		

514.83

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Report # 7-0400-05 Sheet # 002-6 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 2000 OF ISSUE 26 of May 1st, 1956

WEIGHT CHANGES

Fin & Rudder

+ 153.18

	THE W RUDGER				
+	4.22	Fin - Changes to jac	king supports and provision for rudder feel and trim Miscellaneous change to Production Drawings	+	5.87 1.65 4.22
+	4.22				
Ψ×	ont Fugolog	e (Fwd. Sta. 255")			
11	One Puserag	e irwa. Sta. 299"]			
+	7.14	Pilot's Canopy - Deta	ailed estimate to Production Drawings of Canopy and attachments - previously some schemes used.		
+	3.47	Navigator's Canopy -	Installation of attach. bolts to longerons.		
+	0.41 10.51	Windscreen - details Canopy Arches - Entir	rely re-estimated to Production Drawings.		
			Addition of steam outlets	+	3.00
			Addition of switch box	+	0.49
			Detailed estimate of hardware Alterations to some gauges of	+	3.39
			stiffeners etc.	+	3.63
				-	
				+	10.51
+	4.95	Formers F.F Genera	al revisions to angles, doublers etc. on re-issued Production Drawings		
+	5.54	Bulkhead Sta. 255 - 1	Production Drawing estimate including slinging fittings (see also Centre Fuselage)		
+	110.05	Intake Ramp - Fwd. of	Sta. 2018 Structure has been entirely redesigned due to unsatisfactory boundary layer bleed etc.		
8	3.69	Nose U/C Structure -	Pivot bolts etc. for assy. of U/C to structure now with U/C see Landing Gear Group.		
+	14.80	Lower Longerons - Man	y small design changes to Production Drawings since last scheme estimate was made. Lower Longeron Left Hand	+,	7.74
			Lower Longeron Right Hand	+	7.06
4	752 70			-	-CHENNAMING COLCER

Report # 7-0400-05 Sheet # 002-7 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

2.31

2.15

2.47

0.74

7.17

WEIGHT CHANGES TO 7-02 GI ISSUE 26 of May 1st, 1956

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WEIGHT CHANGES

Centre Fuselage (Sta. 225-485)

æ	9.57	Slinging	Fittings	- T	hese	should	l not	have	been	included	in t	he
				C	· F.	since	they	are	part of	Bulkhead	l Sta	255M
				а	ssy.	F.F. a	ind a	re no	w incl	uded ther	°A.	

- 2.02 Radar Access Door General re-estimate to Production Drawings. Reduction in weight of fastener allowance, Kaylocks replace AN Nuts etc.
- 2.26 Duct C.F. Actual weights of Production Aircraft ducts were obtained.
- 3.45 Formers Misc. minor Production Drawing changes.
- 0.79 Sealing Missile Bay Production Drawing estimate
- 7.95 Integral Fuselage Tank Dorsal Angles and Buttstraps. now called up on C.F. Skins

	-	7.95
Misc. Production Drawing changes	+	2.79
Addition of Toggle attach. Stiffeners	+	2.58
now carred up on G.r. Skins	ca	13.32

- 0.60 Dorsal Deflector Shield estimate to Production Drawings 3.19 Fuse attach now with skins 2.59
- 0.60 79.89 C. F. Skins - Completely re-estimated to Production Drawings
- Stringers now called up on skins (see below) 51.73 Skins 469" Aft .04 Al. were Mg. Addition of Access Door Aft Sta. 458" Addition of Wing Side Rib Sealing Clips etc. Addition of Dorsal Angle, Longeron etc.

(see C.F. Tank) + 13.32 Addition of Steel insert aft end Sta. 469, alterations to splices etc. 79.89

- 51.73 Stringers C.F. Report deleted, now called up with C.F. Skins (see above)
- 0.49 Dorsal 268" 317" Mostly to Production Drawings, redesign of aft end.
- 1.49 Dorsal Fairings over Fuel Tank Production Drawing estimates Weight changes due to refinement.

NON CLASSIFIED

Report # 7-0400-05 Sheet # 002-8 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

weight changes to 7-0400 05 ISSUE 26 of May 1st, 1956

WEIGHT CHANGES

Centre Fuselage (Contid.)

+	2.13	Equipment Bay Structure -	Addition of cross Strut Sta. 292"	. + +	0.38 1.33
			Addition of Turbine Mounting Brkt.	+	0.42
				+	2.13
6	4.38	Longerons C.F Production	on Drawing estimates		
	,		Alterations to End Fittings Top Outer Longeron Deletion of hardware allowance	+	0.82
		,	on Lower Longerons - hardware called up on Skins	æ	5.20
				-	4.38
+	3.64	Armament Bay Roof - added	equipment brackets etc.	+	2.40
T	J 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	III mamoria Day 100 an anada	Minor alterations to floor	+	1.24
				+	3.64
+	4.20	Structure below Lower Long	geron - Production Drawing estimates of panels in Radar Bay many small changes		
+	1.45	Bulkhead Sta. 485 * * - 1	Revision to re-issue of Production Drawings Stabilizers at Stringers # 13 & 15	+	3.52
			now on Skins	-	2.07
			44V 18 V44 M6haba4M	+	1.45

^{+ 16.26 *} Although Bulkhead Sta. 485" was previously always considered to be in the C.F. for the Weights Component breakdown, in Summaries up to Issue 26 the Bulkhead was included with the Duct Bay at Stress Office request.

Now, however, it will be transferred to the C.F. as it is called up and assembled with this component. This results in a further apparent increase in the C.F. of + 109.84 lb and a corresponding decrease of -109.84 lb in the Aft Fuselage - in the Weight and C.G. Summary only.

WEIGHT CHANGES TO 7-0400-05 RESUE 26

Report # 7-0400-05 Sheet # 002-9 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

of May 1st, 1956

WEIGHT CHANGES

Aft. Fuselage (Sta. 255 Aft.)

AL U	o raberas	S (OUR & AZZ AZ OO)	
+	7.26	Longerons D.B Production Drawing estimate, alterations to extrusion since last estimate + 4.74 .072" thick plate has been widened + 0.58 Detailed estimate of hardware + 1.94 + 7.26	
+	0.21	Dive Brake Accommodations - Addition of Mounting Brackets to decking.	
+	1.96	Dive Brakes - Re-estimate of main lever and bushing	
œ	0.16	Longitudinal Beams D.B Re-issue Production Drawings fore and aft beam	
+	2.13	Mounting Brackets Side Panels D.B new report to cover misc. mounting brackets for Air-Conditioning etc.	
ca	5.00	Stringers D.B This allowance is now absorbed in the lower panel assy. Weight.	
	24.37	Skin D.B Lower Panels removed and included with lower panel assy 25.64 Misc. changes due to detailed information + 1.27	
æ	69.99	Formers D.B Now side members only, for convenience all lower beams and formers are now included in a separate report	
+	76.06	Formers Lower Panel D.B New report, these formers include some items previously designated on schemes as heat exchanger mountings. Actual increase in formers (20 off) is + 22.39	
+	62.85	Lower Panel D.B New report to include all mounting brackets, intercostals, diaphs. etc. Addition of equipment mounting brackets + 3.14 Redesign of heat exchanger assy. + 8.39 Diaphragms, intercostals, stiff.etc.+ 6.95 Actual Increase + 18.48	

Date: December 1st, 1956 Aircraft: C-105 With J75

P5 Engines as Interim Power Plant Report # 7-0400-05 Sheet # 002-10 Issue 27 Prepared By K. Griffin Checked By: E. Burnett

weight Chynges, to 750400-05 Issue 26

WEIGHT CHANGES

Aft. Fuselage (Cont d)

+	60.83	Duct D.B Production D	rawing estimates have now been made, incorporating many design changes051 Al. aft skin now spliced at Sta. 540.2 was at Sta. 545,35 Addition Splice ring & packing at 540.2 Splice Strap at Sta. 510.4 .04 Al. was .032 Gill Doors redesigned Attach. struts.heavier fork	+ + + +	2.04 4.00 0.47 12.34
			ends etc. Side Torque boxes and attach. beams - redesign end fittings.	+	4.80
			pick-ups etc. Pressure Seal - increase in brackets, retainer CSE772 was	+	14.37
			CSE639, extruded lower seal added etc.	+	6.08
			Addition of Spring Retainers for Door Addition of TStiffs, near side	+	0.91
			torque boxes	+	1.62
			Redesign of aft Torque Box - larger cross section etc. Increase in number of diaphs;	+	8.92
			gussets added etc.	+	5.28
				+	60.83
+	4.55		ed now to Production Drawings previous made to incomplete information on	S	
			splice plates, finger plates etc. Delete hardware allowance now with	+	6.55
			skins	æ	2.00

- + 0.07 Top Longerons E.B. Production Drawing estimate
- + 27.66 Torsion Box E.B. Some redesign, estimates now made to Production Drawings; the previous allowance made was too low.

continued.

4.55

Date: December 1st, 1956 Aircraft: C-105 With J75 P5 Engines as

Report # 7-0400-05 Sheet # 002-11 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 7-0400-05 ISSUE 26 of Way lets 1956

WEIGHT CHANGES

Aft Fuselage (Cont'd.)

+	15.14	Service Access Doors E.B Production Drawing estimates No. 4 panel increased in size Beam I/B Longeron omitted Increase in hinges etc, some now called up here that were	++	20-40
		with Engine Doors Estimate of structure for Doors # 3 & 4 (includes 4.49 lb of structure at press vents # 7 & 8	+	4.55
		See also Tunnel)	++	4.78
+	56.42	Engine Access Doors E.B Engine Access Doors # 1 & 2 have been completely redesigned at R.C.A.F s request. Engine Door # 1	_	2/ 00
		Engine Door # 2		34.00 23.80
		Surround Structure Door # 1	+	4.00
		Hinges now with Torsion boxes etc.		
			+	56.42
+	6.48	Light Formers E.B Production Drawing estimates of formers	+	4.68
	•	Splices at Longerons now included	+	1.80
		1	-	
			+	6.48
+	58.08	Skins E.B Completely estimated to Production Drawings Skin splices & Manifold access		
		hole added	+	8.61
		Skins - more details cut-outs etc.	omo	2.87
		Hardware allowance was too low Pressure Vents added here (see	+	11.54
		Tunnel)	+	21.56
		Engine Mounting Access Doors (See		
		Intermediate Formers)	+	8.77
		Air Bleed Outlet (see Tunnel) Misc. Production Drawing changes	+	6.97
		without it oddedion blawing changes	-	3.50
			+	58.08
+	11.76	Intermediate Formers E.B Redesign and re-estimate of		
		formers 712.34, 717.36 & 663.65 Webs .064 were .04, lower booms		
		.125 were .064 etc.	+	16.59
		Wing Attach. Brackets 712.3 & 717.3		100)/
		now called up with Engine Mtg. Acce	ss	
		Door (see skins)	<u> </u>	4.83
			+	11.76

Report # 7-0400-05 Sheet # 002-12 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

weight changes to 7-0400-05 Issue 26 of May 1st 1956

WEIGHT CHANGES

Aft. Fuselage (Cont d.)				
+	6.85	Longitudinal Members E.B Production Drawing estimates - this item was duplicated in the Tunnel report and allowance in this report of 4.19 lb was out- dated (now deleted from Tunnel)	-	
+	23.48	Heavy Formers E.B Production Drawing estimates I/B Rail Supports called up on former drawings * O/B Rail Supports called up on former drawings * Redesign & Misc. changes to	+	13.27 3.94
		Production Drawings	+++++++++++++++++++++++++++++++++++++++	6.27
		* See also Power Plant Group		
+ .	9.12	Misc. Structure E.B Items of structure on G.A's not readily assigned to existing reports, mounting brackets etc.		
.	2.30	Engine Tunnel E.B Completely estimated to Production Drawings.		
		Increased gauges lower shroud Insulation - more details -	+	13.77
		lighter fasteners	œ	10.08

		Drawings.		
		Increased gauges lower shroud	+	13.77
		Insulation - more details -		
		lighter fasteners	cas	10.08
		Airflow restrictor - cover .025		
		was .032	œ	1.86
		Engine Firewall	+	1.55
		Beam Shroud Att. O/B some gauges		
		.064 were .04 etc.	+	3.48
		Delete Longitudinal Supports		
		(duplicated above)	=	10.54
		Pressure Vents etc. partially in		
		Skins (actual increase per A/C		
		+ 10.21)	6	14.43
		Angle - seal attach. Sta. 742.5		
		added	+	3.65
		Misc. Production Drawing changes		
		hardware etc.	+	12.16
			_	2.30
1.38	Formers Fixed R.F - Form	ner Sta. 803 redesigned	+	3.77
		Production Drawing estimates of		
		Formers 753, 783, 788, 793 & 798	œ	2.39

Report # 7-0400-05 Sheet # 002-13 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

weight changes To 7-0400-05 16 sue 26 of May 180, 196

WEIGHT CHANGES

Aft. Fuselage (Cont d.)

+ 1.30 Tunnel Fixed R.F. - Addition of stops on I/B Rail

+ 15.18 Nacelles R.F. - Completely estimated to Production Drawings incorporating some design changes and increases in former gauges since last scheme drawing estimates were made.

Landing Gear Group:

+	103.38	Main Undercarriage - Wheels increase in Weight (Goodyear) Dowty Increase in leg Assy. Actual Weight Dowty parts by Avro resulted in further increase	+ 7.60 + 86.50 + 9.28 + 103.38
-	284.84	Main U/C Hydraulics - now included with Main Hydraulic System - see Equip. Group	
+	7.04	Main U/C Doors & Fairings - Completely re-estimated to Production Drawings	
+	18.34	Nose U/C - Actual Weights of Jarry supplied parts Actual Weight of Dowty liquid spring assy. Pivot added- previously included in the basic structure	+ 10.55 + 3.13 + 4.66 + 18.34
-	1.38	Nose U/C Deor and Fairing - Fairing links now with U/C Redesign of Fairing end Door addition of angles etc. Jack pick-up fitting on door	- 0.84 - 0.44 + 0.90 - 1.00
	99•55	Nose U/C Hydraulics - now included with Hydraulic Main System - see Equip. Group	

^{- 257.01}

^{+ 346.95}

Report # 7-0400-05 Sheet # 002-14 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO TOADS -05 ISSUE) 26

of May 1st. 158

WEIGHT CHANGES

Power Plant & Services

+	87 . 06	Accessories Gear Box Fuselage - There has been a considerable increase in weights supplied by Sarger previously 150 lb was allowed for the engine installation - this report now contains only that part mounted on the fuselage (237.06 lb). Deareator tanks (6 lb) have been transferred from Oil & Hydraulic Fluid Cooling report. Gear Boxes & Starters on Engine - This report previously only contained the Starters now includes that part of the Gear Boxes which is mounted on the Engine (includes resestimate of piping for Drives 10 lb	it,	
		was previously allowed on Engine)		
=	77.30	Engine Mounts - Items included on Engine now in a separate report (see below) Duplication of rail aft Sta. 742.5 also included in R.F. Tunnel assy. Rail Supports now on Heavy Formers	0 0	55.53 9.60
		(see also Structure Sta. 485" Aft) Production Drawing estimates	+	13.02
			-	77.30
+	50 -42	Engine Mounting Accessories - new report - see above completely re-estimated. Housing assy. centre fitting redesign Engine Brackets (4) gauge decreases Bearings - type changed Roller bolts etc. more details	+ = =	3.36 1.93 3.57 2.97
		Actual change	-	5.11
+	2.27	Engine Anti-icing - Ice detectors & wiring are included in electrics Production Drawing estimates of	.	5.00
		Fairings Production Drawing estimates of Struts	+ +	4.19 3.08
			+	2.27

Report # 7-0400-05 Sheet # 002-15 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 7-0400-05 Issue 26 of May 1st 1956

WEIGHT CHANGES

Power Plant & Services (Cont d.)

			*		
+	152.38	Engine Service Accesso	ries - New report for recording convenience including drives, pipes etc. Oil Breather pipe - no allowance Nozzle Air Ejection primary no allow.	++	14.25
			Systems disconnect bracket and cover Altenator Drives (4.5 lb Oil now in Operational Load, 10 lb piping in Engine Gear Boxes (see above))	+	3.47 14.50
			Actual change	-	
	108 de		· · · · · · · · · · · · · · · · · · ·	+	5.36
-	127.85	Engines - Constant Spe	ed Drives now in Engine Accessories (see above) Addition of Pressure Oil Filler liners were previously installed by Pratt &	-	138.50
			Whitney	+	4.80
			Lower Engine Can attachement - re- design etc.	+	5.85
			Engine Can - increase in stiff. gauges etc.	+	9.46
			Adaptor Ring - Skin decrease .04 to .032 etc. Thermocouples - inclusive with Engine	-	3.30
			Weight Heat Exchanger Exhaust Duct - now part of Engine Assy. was with Oil & Hydrauli	<u> </u>	10.00
			Fluid Cooling (see Equip. Group)	+	6.07
			BracketsFireCan Support - no allowance Fairing - Production Drawing Estimate Oil Filler & Pressure Switch - with	+	2.52 0.38
			accessories (see above)	=	8.52
			Misc. changes packing etc. Air-Bleed outlet - that part of assy. on Engine added, was previously on	æ	0.39
	a .		Tunnel	+	3.78
.1.	70 10	Fuel Crates Addition	of Hart Broken and Contact T. W. J.	co	127.85
+	78.49	ruel System - Addition	of Heat Exchanger & Control Valves (see Oil & Hydraulic Fluid Cooling)	+	70.90
			Addition of Mtg. Casting for H/Exch.	+	
			Revised Weight of Fuselage Bag Tanks	+	11.32
			Capacitor Units - actual weights Condensate drains - revised estimate	+	
			Piping etc. excluding H/Exch. inst.	+	1.47 15.31
			Piping etc. H/Exch. inst. added (22.5 lb. was allowed in Oil & Hyd. Fluid		± / 0 /d
			Cooling - equip.)	+	7.09
-	0/0/2	e e	Flow Proportioners Unit - Mnfg.& SpecWt	5 -	5.93
+	263.65			+	78.49

Rep She She Pre Lant Che WEIGHT CHANGES TO 7-048 -05 ISSUE 26 of May 1st, 1936

Report # 7-0400-05 Sheet # 002-16 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES

Flying Controls Group

	riging controls Group				
+	65.00	Mechanical Flying Controls - Steel links replace Al. links for Elevator, Aileron & Rudder controls, due to difficulties en- countered with bearings & expansion problems.			
•	41.00	Dive Brake Hydraulics - deleted from this group, now Integral part of Utilities System (see Equip.)			
0	12.01	F/C Hydraulics I/Wing - Elevator jacks, previous estimate was based on prototype jacks General piping changes etc.	8	10.98 1.03	
	4.07	Rudder Hydraulics - Jack support bracket with Fin Jack & Linkage - actual weight Piping etc Production Drawing estimate	+	3.76 2.58 2.89	
æ	16,20	F/C Hydraulics O/Wing - Aileron jack attach. bracket duplicated - already in O/W Structure Filters & fluid - too high allowance Jacks - actual weight Misc. re-estimate to Production Drawings	60 60 60	7.25 5.12 2.48 1.35	
+	63.79	F/C Hydraulics Main System - Heat Exchangers added were with Oil & Hydraulic Fluid Cooling equip. group. Filters - Actual weights, allowance too low Ground Service Connectors - Actual Weight Pumps, compensators etc. Spec. Weights of Servos incorp. Miscellaneous changes added piping D.B.,	+ + + +	11.30 15.22 3.78 4.77 3.50	
-	108.00	flex hose replaces Al. tube etc. etc.		25.22 63.79	
~	108.00	Auto Pilot & Tie-in - Forms an integral part of the R.C.A.			

Forms an integral part of the R.C.A. System and will in future be considered part of the Equipment Group see note # in that Section.

Date: December 1st, 1956 Aircraft: C-105 With J75 P.5 Engines as

Report # 7-0400-05 Sheet # 002-17 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 7 CLOS-05 ISSUE 26 of May 1st, 1996

WEIGHT CHANGES

Equipment Group

Eq	uipment Gr	<u>oup</u>		
+	3.22	Intake De-icing - Increase due t	o redesign of Intake Ramp.	
-	0.16	Piping et	35 7 0 1 7 0	+ 2.50 + 0.92 = 0.58 = 3.00
+	20.00	Alternator System - This is a 30 increase an emergen	is due to addition of	
+	7.61	Plumbing of flexible to Miscelland place show	ight of Sequence Valves now Steel was Al. & tubing. eous changes links re- ck absorbers etc.	+ 1.75 + 0.94 + 2.42 + 0.15 + 2.35 + 7.61
+	3.78	Windscreen De-misting - re-estime to cable w		
æ	12.30	Fin Pitot System - Weights now in and low pr	ncluded in Fin Structure ressure pneumatics.	
+	22.41	Addition o	of Diff. Pressure Switch of Pitot lines, fin etc. Indscreen press. seal	+ 15.68 + 2.41 + 7.28 - 2.96 + 22.41
+	4.93	Air Conditioning - Stress approve now estima		
	108.00	systems an report. Fuel Oil E System Control Va Oil in Exc	Various items are being with their respective and are removed from this exchangers now in Fuel lives now in Fuel System = hangers now in Op. Load = tanks = see Engine Group = continued	13.30 20.25 6.00

Date: December 1st, 1956 Aircraft: C-105 With J75 P.5 Engines as

Report # 7-0400-05 Sheet # D02-18 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 7 OLDO=05 PSSUE 26 of May Is 1956

WEIGHT CHANGES

Equipment Group(Cont d.)

Oil & Hydraulic Fluid Cooling (Cont	d.)
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H/Exchanger Mounting allowance - see D. B. Structure 2.90 H/Exchanger Ducts - see Engine Group 5.00 4 Air Oil Exchangers now with Flying Control Hydraulics & Main System, allowances were however too low and remaining 4 show an increase in weight over previous allowance for 8 0.60 Piping Mountings etc. absorbed in various other reports 35.15 108.00

- 231.00 Utility Hydraulics This report is deleted and a new set of reports instituted (see below) incorporate Dive Brake, Main & Nose U/C Hydraulics as an integral part of Main Utilities System.
- + 85.35 Utility Hydraulics F.F. New report see Introduction for explanation of weight changes (see also Landing Gear Group)
- + 43.33 Utility Hydraulics C.F. New report See above
- + 228.29 Utility Hydraulics D.B. New report See above
- + 73.28 Utility Hydraulics E.B. New report See above
- + 158.11 Utility Hydraulics I/Wing New report See above
 - A Weight Breakdown has now been received from R.C.A. quoting preliminary weights of the Astra I System.

 This will merely be sub-divided into fixed and removable equipment no attempt will be made at this stage to further sub-divide into various sub-systems, except for the Missile Auxiliaries to be housed within the Sparrow package, hence the following cancellation of detailed Radar reports temporarily until more information is available.
- + 1,078.60 Radio & Radar Removable incorporates all units for the entire R.C.A. Astra I System, except the Missile Auxiliaries in the Missile Package.

continued.

Report # 7-0400-05 Sheet # 002-19 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

55.41

118.35

WEIGHT CHANGES TO 7 0400 05 ISSUE 26 of May 1sc, 1956

WEIGHT CHANGES

Equipment Group(Cont 1 d.)

+	62.94	Radio & Radar Fixed - Based on the weight breakdown from R.C.A. there is a weight decrease of Addition of Antennae not included in R.C.A. weight +
•	329.00	Radar Power Supplies - Report deleted - see note * above
	48.00	Radio Removable - Report deleted - see note * above
=	15.00	Interphone - Report deleted - see note * above
-	65.20	ARN 6 Compass - Report deleted - see note * above
-	119.70	I.F.F. = Report deleted = see note * above
	70.00	OMNI-DME - Report deleted - see note * above
ca	68.00	Data Link - Report deleted - see note * above
-	15.50	Flare-out Altimeter - Report deleted - see note x above
a	48.00	Electronic Missile Control - Report deleted - see note * above
+	3.14	Electrics for Sparrow Pack - New report for circuits in the Aircraft to the pack disconnects and pertaining to it.
8	50.00	Sparrow Pack Structure - Completely re-estimated, pre- viously only an allowance was made (Pack sealing absorbs 106 lb of the weight included here)
+	300.32	Sparrow Pack Mechanisms - entirely re-estimated, previous figure was known to be unrealistic and was based on 60° launchers etc., these are now 150° long and mechanism has been entirely redesigned.
0	155.46	Sparrow Pack Hydraulics - System completely redesigned since last estimate made. Previous system included 2 accumulators at 100 lb each that are now obselete etc.
+	64.10	Sparrow Pack Electrics - New report for switches, relays, cable etc. installed within the pack.

Continued.

Report # 7-0400-05 Sheet # 002-20 Issue 27 Prepared By: K. Griffin Checked By: E. Burnett

WEIGHT CHANGES TO 240400-05 ISSUE 26 of May 1st 1956

WEIGHT CHANGES

Equipment Group (Cont d.)

+ 128.00 Sparrow Pack Electronics - R.C.A. propose to house 178 lb of Missile auxiliaries & installation within the pack, previously only 50 lb of cable was allowed here.

+ 100.00 Surface Finish - 1 coat etch primer, 1 coat chromate primer and 2 coats finish, plus allowance for skin filler.

+ 1,052.09

Operational Load:

+ 34.02 Oil Usable - The trapped and usable oil for the Engine Oil

System and the Gear Box Oil System
has been completely re-estimated.

Only allowances were previously
carried, some in Oil Hydraulic Fluid
Cooling (Equip. Group) some in Engine
Installation with Drives (Engine Group)
etc. Weight removed from other groups
34.50 lb.

- + 11.29 Oil Trapped See note above. Weight change per Aircraft for Total Oil System = (34.02 + 11.29 34.50 = + 10.81 1b.)
- + 125.00 Water For Air-Conditioning Now a requirement for all missions.

 Figure quoted is for Mach. 2.
- + 170.31
- + 2,211.99 TOTAL WEIGHT CHANGES