QCX Avro CF105 R.7-0400-34 Iss-3

# FILE IN VAULT

C-105 With PS 13 Report # 7-0400-34
Engines Issue 3
WEIGHT SUMMARY & C.G. POSITION
N.A.E. Dec. 1/56

National Research Council Conseil national de recherches Canada Canada I.C.1.S.T. C.I.S.T.I. Aeronautical and Bibliotèque Mechanical d'aéronautique Engineering Library et de génie mécanique JAN 0 7 1993 QT Report no.: \_ QCX - AVRO - CF105 - R-7-0400-34- Iss.3 downgraded to: has been XX de-classified by (Name): Michel W. Drapeau (Dept.): A/DND Coordinator, Access to Information Date: Dec. 7, 1992 R. Augus Classification cancelled / changed to: <u>UNCLASSSIFIED</u> By authority of: DRDA 7/DARFT 5-8/DAS Eng 6-4-5 Date: 5 Nov 1992 Signature: B. auligay Unit / Rank / Appointment: DSIS 3, Secretary CRAD HQ DRP Date: December 1st, 1956 Aircraft: C-105 With PS 13 Engines.



# INTRODUCTION

The following is a Weight and C.G. Summary of the C-105 Aircraft with P.S. 13 Engines. Of necessity the figures quoted herein are only of a very preliminary nature.

Information herein is based on Report # 7-0400-05 Issue 27 of December 1st, 1956, Weight Summary for C-105 Aircraft with J75 P5 Engines as Interim Power Plant, with relevant changes made.

#### GENERAL:

- Orenda PS 13 Engines comprise the Power Plant (4,500 lb each). (a)
- (b) A package containing 4 "semi-submerged" Sparrow II Missiles (432 lb each) is currently carried.
- The R.C.A. Radar Astra I is installed weight information being to the latest breakdown received from them, dated October 15th, 1956. (Total Weight of Astra I including Avro installed Antennae = 2,731.5 + 118.5 = 2,850 lb.)

# 1. STRUCTURE

With the introduction of PS 13 Engines the lines of the Rear Fuselage Aft Sta. 742.5% undergo changes, there will be alterations to the Engine Bay Structure, Intake Ramp, Intake Ducts and numerous other changes, however, until more information is available the Structure weight will temporarily be retained as in Report # 7-0400-05 Issue 27.

#### 2. LANDING GEAR

This will remain as in Report # 7-0400-05 Issue 27

#### 3. POWER PLANT & SERVICES

All items in this group will alter for PS 13 installation, however, until more details are available all accessories, mounts, controls, etc. will be retained weightwise as for J75 P5 Installation, the Engine Weight and Alternator Drive only being amended. The fact that the engine rails are now fixed in the Aircraft will be more than compensated for weightwise by there being no additional Fire Can. The PS 13 at 4,500 lb already includes a Fire Can.

Engines - J75 P5 were 5,950 lb each, PS 13 quoted at 4,500 each, therefore, compared to Report # 7-0400-05 Issue 27 Weight Change is - 2,900 Engine Accessories - 40 KVA Alternators replace 30 KVA's resulting 23 in a weight increase on Drives of - 2,877 lb.

#### WEIGHT CHANGE POWER PLANT GROUP

4. FLYING CONTROLS GROUP

This will remain as in Report # 7-0400-05 Issue

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Date: December 1st, 1956 Aircraft: C-105 With PS 13 Engines

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Report # 7-0400-34

Report # 2 Issue 3

Rv.: K. G

Prepared By: K. Griffin Checked By: E. Burnett

## INTRODUCTION

## 5. EQUIPMENT GROUP

The majority of items remain as in Report # 7-0400-05 Issue 27. However, there are some changes relative to that report, as listed below:-

Air-Conditioning - Increases are anticipated here due to redesign of the system resulting from reduced engine bleed pressures and increased Air-Conditioning requirements for Astra I not already catered for in Report # 7-0400-05 Issue 27.

Preliminary figures indicate

+ 215 1b

Alternator System - 40 KVA Alternators will probably be installed on the aircraft with PS 13 Engines replacing 30 KVA Alternators on J75 Aircraft.

110 lb

WEIGHT CHANGE EQUIPMENT

-325 lb

## 6. OPERATIONAL LOAD

All items as in Report # 7-0400-05 Issue 27 with the following exception:-

Water for Air-Conditioning - requirement for all missions Max. Weight at Mach 2 is 260 lb, with J75 Engines weight of water was 125 1b, thus increase over Report # 7-0400-05 Issue 27 + 135 1b.

(All changes relative to Report # 7-0400-05 Issue 27 - C-105 Aircraft SUMMARY With J75 Engines as Interim Power Plant)

WEIGHT CHANGE - AIRCRAFT WEIGHT EMPTY

Power Plant = 2,877 lb + 325 lb. Equipment - 2.552 lb.

WEIGHT CHANGE - OPERATIONAL TOAD (Less Fuel) Water for Air-Conditioning + 135 lb.

WEIGHT CHANGE - OPERATIONAL WEIGHT EMPTY - (A/C Less Fuel)

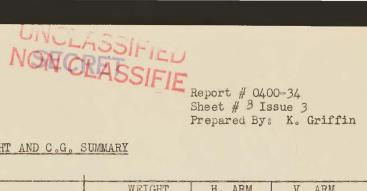
Report # 7-0400-05 Issue 27

Report # 7-0400-34 Issue 3

46,528

= -2,417 lb





Date: December 1st, 1956 Aircraft: C-105 With PS 13 Engines

# WEIGHT AND C.G. SUMMARY

DESCRIPTION	WEIGHT lb.	H. ARM ins.	V. ARM ins.
STRUCTURE	18,087.08	565.39	137.29
Wing	9,989.07	643.60	141.93
Fin and Rudder	999.96	754.27	201.96
Fuselage Struct. Fwd. Sta. 255"	2,390.47	187.41	129.61
Sta. 255" - 485"	1,672.37	378.59	129.82
Sta. 485 Aft.	3,035.21	646.37	110.90
UNDERCARRIAGE - Up Position	2,604.33	488.72	134.65
Main Undercarriage	1,951.62	539.57	141.00
Main U/C Doors & Fairings	294.36	539.29	136.01
Nose Undercarriage	333.81	170.80	99:70
Nose U/C Doors & Fairings	24.54	162.24	88.23
POWER PLANT & SERVICES	10,841.50	672.83	120.34
Engines PS 13	9,345.00	688.33	119.74
Gear Box & Drives on Fuselage	237.06	603.73	104.00
Engine Centrols	29.19	377.46	118.91
Gear Box & Starter on Engines	150.18	610.55	96.37
Engine De-icing	70.37	562.80	115.09
Fire Extinguishing System	70.52	701.99	127.72
Engine Mountings	189.19	633.40	127.82
Fuel System	749.99	542.97	135.70
FLYING CONTROLS GROUP	1,677.82	688.36	139.58
Mechanical Flying Controls	904.40	689.05	147.08
Flying Control Hydraulics	773.42	687.56	130.81
EQUIPMENT - FIXED & REMOVABLE	8,072.95	319.10	111.18
Instruments	53.30	153.98	140.27
Probe	23.00	9.74	108.00
Cockpit Pressure Sealing	5.00	186.00	130.00
Oxygen System	43.44	227.72	142.18
Air-Conditioning System	927.68	327.16	132.46
Hydraulic Main System	588.36	501.22	117.38
Cabin Insulation	11.91	179.24	130.00
Brake Parachute	69.69	784.88	131.17
Electrical System	1,068.34	418.20	116.16
Low Pressure Pneumatics	39.01	478.47	127.28
Oil & Hydraulic Fluid Cooling	22.00	579.50	92.00
Intake De-icing	85.84	206.52	118.79 126.04
Radome Anti-icing	23.46 54.41	222.04	154.40
Canopy Actuation	20.65	117.37	125.23
Cabin Consoles	10.00	268.00	95.00
Radar Door Actuation Ejector Seats	186.00	201.10	136.25
Radio & Radar - Removable	2,001,80	178.02	103.53
Radio & Radar - Removable Radio & Radar - Fixed	CS 671 35	230.67	108.98
Sparrow Pack Structure	850.00	390.84	96.00
Sparrow Pack Mechanisms	625.32	376.67	99.22
Sparrow Pack Hydraulics	350.19	368.83	99.00
		continued	

Report # 7-0400-34 Sheet # 4 Issue 3 Prepared By: K. Griffin

11

# WEIGHT AND C.G. SUMMARY

Contraction and Contraction of the Contraction of t	WEIGHT	H. ARM	V. ARM	C.G. POSITION
DESCRIPTION	lb.	ins	ins.	%. M.A.C.
Equipment - Fixed & Removable - Cont d.				
Sparrow Pack Electronics	178.00	332.00	100.00	
Sparrow Pack Electrics	64.10	362.29	95.00	
Surface Finish	100.00	591.52	140.20	
AIRCRAFT WEIGHT EMPTY	41,283.68	545.60	127.66	
USEFUL LOAD	17,912.18	512.76	136.17	
Crew	430.00	194.00	136.50	
Oil	130.39	609.19	117.17	
Alcohol - Radome De-icing	22.00	93.00	138.00	
Engine Fire Extinguishing Fluid	25.00	730.00	129.00	
Residual Fuel	218.40	553.98	134.04	
Missiles	1,728.00	389.29	88.30	
Oxygen Charge	13.39	259.69	159.91	
Water for Air-Conditioning	260.00	268.00	132.00	
Fuel for Combat Mission	15,085.00	539.26	141.90	
U/C Up		535.66	130.24	27.53
Normal Combat Mission	59,195.86			
U/C Down		537.30	127.52	27.99
Half Combat Mission Fuel	7,542.00	540.98	144.27	
967 Gals. at 7.8 lb/gal.				
U/C Up		535.39	128.88	27.46
Combat Weight (Half Combat	51,652.86			
Mission Fuel) U/C Down		537.27	125.76	27.98
U/C Up		534.43	126.25	27.19
Operational Weight Empty	44,110.86			
U/C Down		536.64	122.60	27.80
U/C Up	AND SECURITY CONTRACTOR AND	540.35	127.80	28.83
Operational Weight Empty	42,382.86			
(less Missiles) U/C Down		542.65	124.00	29.46
Max. Internal Fuel	19,843.00	538.88	144.32	
2544 gal. @ 7.8 lb/gal.	_,,,0,,,000	),,,,,,	-44070	
U/C Up		535.81	131.86	27.58
A.U.W. Max. Internal Fuel	63,953.86	232.01	131.00	21000
U/C Down	0000000	537.33	129.34	27.99
	00	751000	127074	~1077
Max. External Fuel 500 gal.	1 226 00	522 2/	60.64	
at 7.8 lb/gal. + Drop Tank	4,226.00	522.34	The second secon	07 05
U/C Up	(0.200.0/	534.98	127.44	27.35
A.U.W. Max. Internal and	68,179.86	526 17	125 00	27 71
External Fuel U/C Down		536.41	125.08	27.74

N.B. 1) Aircraft Datum = 120% above nominal ground line

1) Aircraft Datum = 120" above nominal grown

2) \* Together make R.C.A's Weight of Astra I + 118 by of AVRO supplied Antennae.

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