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PROPERTY SYSTEM

Section \$627

FIRE PROTECTION

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J. H. PARKIN BRANCH

MAY 24 1995

ANNEXE J. H. PARKIN CNRC - ICIST

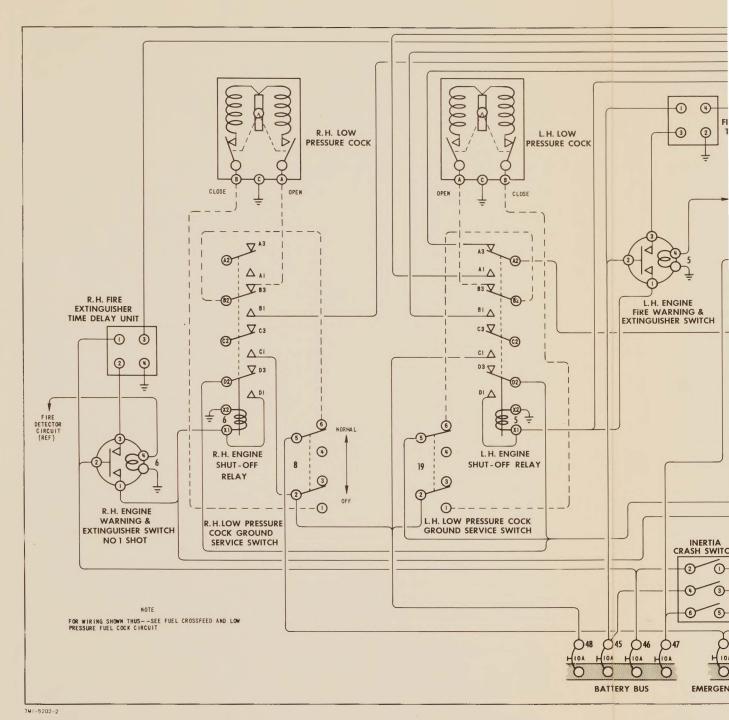
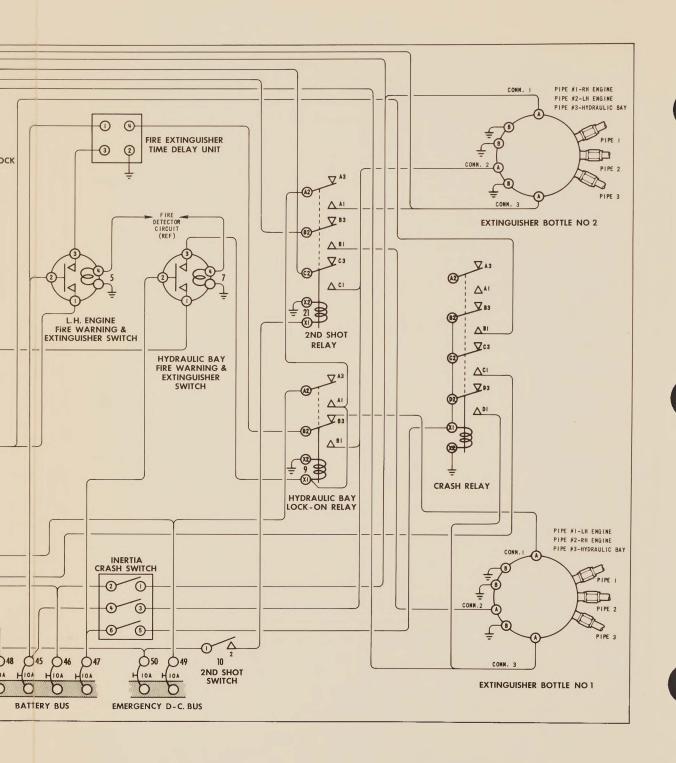
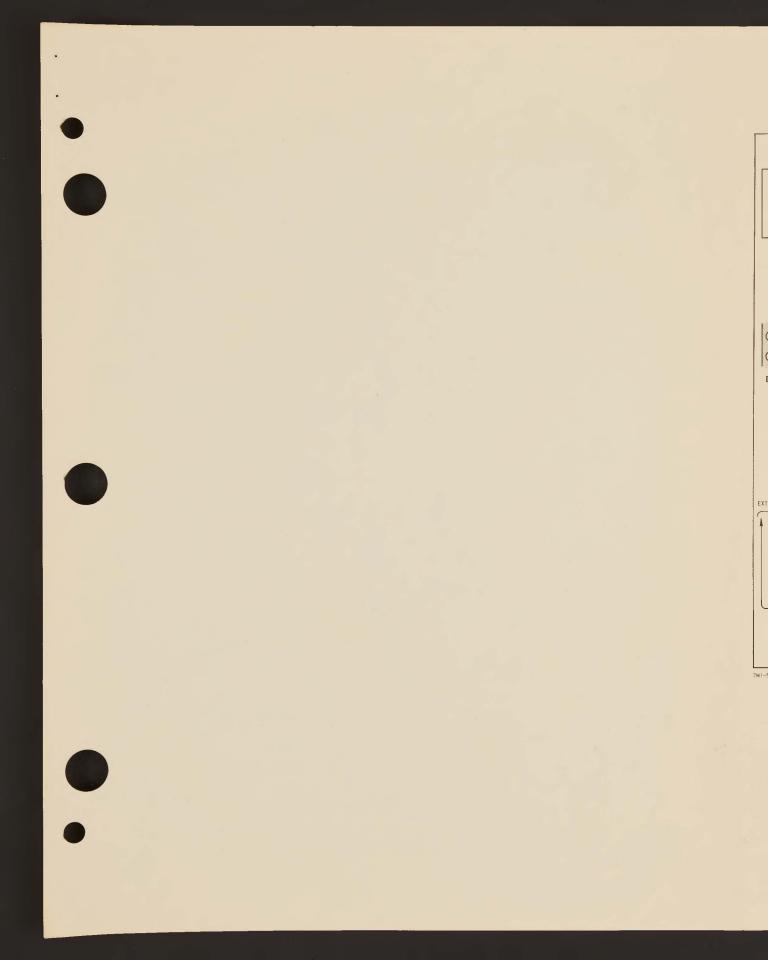


FIG. 2 FIRE EXTINGUISHER ACTUATION CIRCUITS





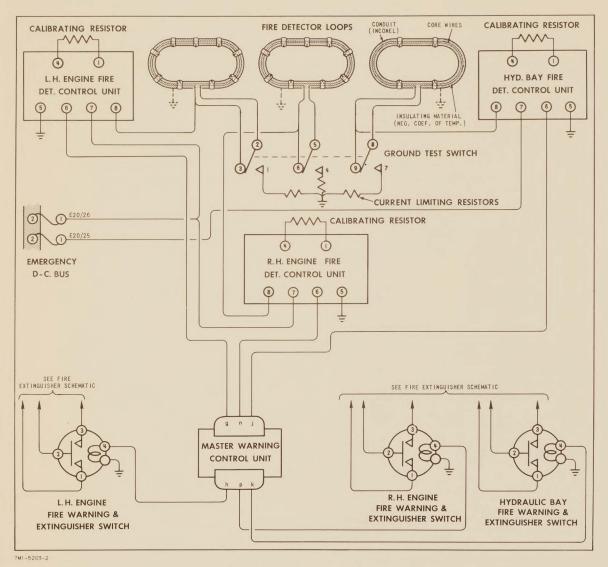


FIG. 1 FIRE DETECTION CIRCUITS

SYSTEM DATA SHEET

SYSTEM	SUB-SYSTEM	AIRCRAFT EFF'TY	REF. NO.
ELECTRICAL	FIRE PROTECTION	25201	11-12

DESCRIPTION

General

- 1. The fire protection system is comprised of fire detection and fire extinguisher actuation circuits.
- 2. The fire detection circuits detect and indicate overheating conditions in the LH engine compartment, the RH engine compartment and the hydraulic bay which is the section of the fuselage between the engine compartments. In addition to the hydraulic equipment, fuel system and electrical system components are located in this section.
- 3. In the extinguisher actuation circuits, two fire extinguishers are fitted which can be discharged individually into any two of the three fire detection areas. Alternatively, both extinguishers can be discharged into any one of the three areas, or both extinguishers can be discharged simultaneously by an inertia switch. In this case, the content of one extinguisher is discharged into both the LH and RH engine compartments, and the content of the remaining extinguisher is discharged into the hydraulic bay.

Fire Detection Circuits

- 4. The three fire detection circuits are electrically identical. Each circuit is comprised of a number of lengths of a specially constructed heat detector conduit, a detector unit and a fire warning indicator light which, when depressed, actuates the corresponding actuation circuit of the extinguishers.
- 5. The detector conduit consists of two uninsulated wires embedded in an insulating material which has a negative co-efficient of temperature, i.e. a rise in temperature results in a drop in the insulation resistance value. The wires and the insulating material are contained in a conductive sheathing of Inconel which is grounded at intervals to the aircraft structure. One of the core wires is connected to the sheathing, and the other is connected to one end of the coil of a relay located in the detector unit.
- 6. The other end of the relay coil receives a d-c supply from the emergency d-c bus. If an overheat condition occurs, the leakage current across the core wires at a pre-determined temperature will be sufficient to energize the relay. The relay, when energized, completes a supply circuit from the emergency d-c bus to the fire warning indicator located in the cockpit.
- 7. The lengths of detector conduit can be constructed to have different operating limits by increasing or decreasing the leakage current flow for any given temperature. This permits a complete circuit to indicate overheating conditions at different temperatures along its length. The detector circuits of the engine compartments are assembled in this manner.
- 8. The conduit lengths fitted forward of the engine firewall are constructed to permit sufficient leakage current to flow and energize the detector relay if the ambient

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temperature exceeds 260°C (500°F), or if at least six inches of the conduit is subjected to 315°C (600°F). The conduit lengths fitted aft of the firewall are constructed to permit sufficient leakage current to flow and energize the relay if the ambient temperature exceeds 307°C (585°F), or if at least six inches of the conduit is subjected to 380°C (715°F). The detector conduits in the hydraulic bay are constructed to permit sufficient leakage current to flow and energize the relay if the ambient temperature exceeds 232°C (450°F), or if at least six inches of the conduit is subjected to 288°C (550°F).

- 9. A resistor fitted externally on each detector control unit facilitates calibration of the relevant system to the required temperature operating limits. These calibrating resistors are, in effect, in parallel with the detector conduit core wires and provide a small, standing current through the relay coil of the detector unit. By altering the ohmic value of the calibrating resistors, the leakage current flow between the core wires required to energize the corresponding detector unit relay will be increased or decreased. Increasing the ohmic value of the calibrating resistors, with respect to the core wire resistance, will increase the operating temperature limit. Conversely, decreasing the ohmic value will decrease the operating temperature limit.
- 10. As the detector circuits are formed in loops, they will continue to operate normally despite a break occuring in the live core wire. To facilitate checking this core wire for continuity, a switch is provided to interrupt the loop and ground the core wire. This simulates a leakage current to ground and continuity of the wire will be indicated if the relevant fire warning lights illuminate. The switch is a three-pole type and grounds all three detection circuits simultaneously. To restrict the current flow when the detector circuits are grounded, a current limiting resistor for each circuit is inserted between the test contacts of the switch and ground.

Extinguisher Actuation Circuits

- ll. Two fire extinguisher actuation circuits, namely a first shot circuit and a second shot circuit, are provided for each one of the three fire detection areas. The first shot circuits are actuated by depressing the relevant fire warning indicator. The second shot circuits which are inoperative until the corresponding first shot circuits have been selected, are actuated by operating a second shot switch.
- 12. Due to the fact that the first shot circuit of the LH engine compartment and the hydraulic bay discharge the same extinguisher, the operation of these circuits is interrelated. When the fire warning indicator of the LH engine is depressed, a supply circuit is completed to an engine shut-off relay and a time delay unit. The shut-off relay, when energized, does the following:
- (a) Completes a self-holding circuit which holds the relay in the energized position until the supply circuit is interrupted, i.e. selecting the Master Power switch to OFF.
- (b) Completes a preparatory supply circuit to the relay-open contact of a second shot relay.
- (c) Transfers the low pressure fuel cock supply circuit from the open side to the close side.
- (d) Transfers the hydraulic bay first shot circuit to the remaining extinguisher.

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SYSTEM DATA SHEET

SYSTEM	SUB-SYSTEM	AIRCRAFT EFF'TY	REF. NO.
ELECTRICAL	FIRE PROTECTION	25201	11-12

- 13. The time delay unit completes the supply circuit to the relevant extinguisher after a delay of two and one-half seconds. This delay ensures that the low pressure fuel cock will be fully closed before the extinguisher discharges. The supply to the extinguisher is maintained by the delay unit for a period of four seconds irrespective of the time during which the fire warning light is kept depressed. The circuit from the time delay unit to the extinguisher is completed via the relay open contacts of a lock-on relay in the hydraulic bay extinguisher actuation circuit. If the hydraulic bay circuit has been actuated prior to the LH engine circuit, the lock-on relay will be energized which transfers the supply through the relay closed contacts to the remaining extinguisher.
- 14. The hydraulic bay extinguisher circuit, when actuated by depressing the hydraulic bay fire warning indicator, completes a supply circuit to a lock-on relay and, via the relay-open contacts of the LH engine shut-off relay, to the relevant extinguisher. If the LH engine extinguisher circuit has been actuated prior to the hydraulic bay, this shut-off relay will be energized which transfers the supply through the relay closed contacts to the remaining extinguisher. The hydraulic bay lock-on relay, when energized, completes a self-holding circuit and a second shot preparatory circuit and transfers the LH engine first shot circuit to the remaining extinguisher.
- 15. An engine shut-off relay and a time delay unit are incorporated in the RH engine extinguisher circuit. When the RH engine fire warning light is depressed, a supply circuit is completed to the shut-off relay and the time delay unit. The shut-off relay, when energized, completes a self-holding circuit and a second shot preparatory circuit and transfers the RH low pressure fuel cock supply to the close side.
- 16. The time delay unit, after a delay of two and one-half seconds, completes a supply directly to the extinguisher for a period of four seconds irrespective of the time during which the fire warning light is kept depressed.
- 17. The second shot circuits are identical in operation. Operating the 2nd SHOT switch will complete a supply circuit to the second shot relay. This relay, when energized, completes all three second shot circuits, but only one will be live due to the operation of the first shot circuit.
- 18. The inertia switch is paralleled into the second shot circuit of the LH engine and the first shot circuit of the RH engine and the hydraulic bay. Therefore, when the inertia switch is tripped by a deceleration force in excess of 6G, it completes a supply circuit from the battery bus which discharges both extinguishers. The second shot circuit of the LH engine and the first shot circuit of the RH engine discharge the same extinguisher into the LH and RH engine compartments. The first shot oircuit of the hydraulic bay discharges the remaining extinguisher into the hydraulic bay. In addition to actuating the extinguisher circuits, the inertia switch energizes a relay which completes a supply circuit to the LH and RH engine shut-off relays. This action closes the low pressure fuel cocks.

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SYSTEM		SUB-SYSTEM		COMPO		REF. NO.
ELECTRICAL		FIRE PROTECTION	ī	Fire Warning I Extinguisher F - 3	ush Switches	11-12-1
AVRO PART NO. CS-S-155		MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT E	
OVERHAUL LIFE:	KNOWI	V-	ES	TIMATED- 1500 h	ours	
FUNCTION	actuati areas,	e as a combined fire on switch. One assen viz.LH engine compart ic Bay.	bly for	each of the thre	e detection	ner
LOCATION	Front c	ockpit LH console, pa	nel El4.			
ACCESS					МЕ	N X MINUTES
	LH cons	ole - four quick-fast	eners.			
REPLACEMENT PRO	Fit and lock-wa	secure assembly to pasher and nut provided circuit wires to lugs anel in console - fou	l.		ME.	N X MINUTES

INSPECTI	ON				•			1	MEN X	MINUTES
			on the Check and pr Depres	that the is panel. that the coperly so is the indicating return	circuit wi ldered.	res are se	ck that			
FUNCTIO	NAL CHECK	s						1	MEN X	MINUTES
· ·										
GINCOIND	HANDLING	AND GIVE	OND TEST	EQUIT WEN						
SPECIAL	TOOLS TO	REMOVE O	R SERVICE							
REMARKS	5									
ISSUE	1									
DATE	4 Feb 57									

COMPONENT DATA SHEET

SYSTEM		SUB-SYSTEM		COMPOR	NENT	REF. NO.
ELECTRICAL		FIRE PROTECTION		Fire Detecti Units -		11-12-2
AVRO PART NO.		MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT E	FFECTIVITY
7-1154-18		Walter Kidde	P/N	871510	25203	
OVERHAUL LIFE:	KNOWI	N-	ES	TIMATED- 350	hours	
	condition	lete a supply circuit ons are detected. Th tion with their corre compartment and Hydra	ree unit	s are fitted whi LH engine compa	ch operate in rtment, RH	
		cessory Panel E5, loc the Missile Bay.	ated on	bulkhead at stat	ion	
ACCESS					ME	EN X MINUTES
		ward mounts by removi				
REPLACEMENT PRO	Fit and Connect	secure unit to panel and secure circuit w nd position panel, se	iring to	terminals.	МЕ	EN X MINUTES

TMI-3413-2+5

INSPECTION		MEN X MINUTES
	Check that the unit is securely mounted. Check that the circuit wiring is securely and properly connected.	
FUNCTIONAL CHECKS		MEN X MINUTES
GROUND HANDLING AND G	ROUND TEST EQUIPMENT	
SPECIAL TOOLS TO REMOVE	OR SERVICE	
REMARKS		
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SYSTEM	SUB-SYSTEM		COMPON Fire Protec	·-·	REF. NO.
ELECTRICAL	FIRE PROTECTIO	N	(Portion forwar RH engine fi	rd of LH and	11-12-3
AVRO PART NO.	MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT E	FFECTIVITY
7-1195-29 (3 off) 7-1195-27 (1 off)	Walter Kidde			2520	1
OVERHAUL LIFE: KN	OWN-	ES	TIMATED- 1500	hours	
FUNCTION To d or 1	etect ambient temperatur calized heating in exce	es in exess of 31	cess of 260°C (50°C (600°F).	00 ⁰ F)	
	ne compartment forward o			•	
ACCESS When	engine is removed, unob	structed		ME	N X MINUTES
ends	ect and secure the elect of the length of condui				N X MINUTES
Fit	edjacent lengths. and secure the length of cture with the fixed sec				
781-3913-2-9					

INSPECTI	ON				I Hall a			МЕ	N X	MINUTES
			urely length Check	and prope of the 1 that the ghout its	rly fitted oop. loop is fi	connector throughout tted secure the aircr	relv.	-		
FUNCTIO	NAL CHECK	(S						МЕ	NX	MINUTES
GROUND	HANDLING	AND GRO	OND LEST	EQUIPMEN						
SPECIAL	TOOLS TO	REMOVE O	R SERVICE							
REMARKS	6				-					
						*				
ISSUE	1									
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SYSTEM	SUB-SYSTEM		COMPON		REF. NO.
ELECTRICAL	FIRE PROTECTION	4	Fire Protect (Portion aft of engine fir	LH and RH	11-12-4
AVRO PART NO.	MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT E	FFECTIVITY
7-1158-77 (1 off) 7-1158-71 (1 off)	Walter Kidde			252	01
OVERHAUL LIFE: KN	-NWC	ES	TIMATED- 1500	hours	
FUNCTION To de or lo	tect ambient temperature calized heating in exces	es in exc ss of 380	ess of 307°C (58°C (715°F).	5 ^o F)	
LOCATION Engir Detec	ne compartment aft of fint tor length 28 feet, asse	rewall (I embled fr	H and RH) om 2 lengths.		
ACCESS				МЕ	N X MINUTES
REPLACEMENT PROCEDU	JRE			ME	N X MINUTES
ends the a Fit a	oct and secure the elect of the length of condui- djacent lengths. and secure the length of cture with the fixed secu	t to the conduit	connectors of to the aircraft		

INSPECTION					MEN :	X MINUTES
	Check that the ele securely and prope length of the loop Check that the loo throughout its len structure.	rly fitted o p is fitte	throughou	it the		
FUNCTIONAL CHECKS					MEN :	X MINUTES
GROUND HANDLING AND GRO	UND TEST EQUIPMEN	IT				
SPECIAL TOOLS TO REMOVE O	R SERVICE					
STEGINE TOOLS TO KEMOVE O	N SERVICE					
REMARKS						
REMARKS						
ISSUE 1						
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SYSTEM		SUB-SYSTEM		COMPON	NENT	REF. NO.
ELECTRICAL		FIRE PROTECTION	ИС	Fire Prote (Hydrau)	ection Loop Lic Bay)	11-12-5
AVRO PART NO.		MANUFACTURER	MAN'F	R'S PART NO.	AIRCRAFT E	FFECTIVITY
7-1150-5027 (8 off) 7-1150-5029 (1 off)		Walter Kidde			2520	1
OVERHAUL LIFE:	KNOWI	N-	ES ⁻	TIMATED- 1500	hours	
FUNCTION To	detec	et ambient temperature ized heating in exces	es in exc ss of 288	ess of 232°C (44°C) (550°F).	50°F)	
		c Bay. length 85 feet, asse	embled fr	om nine lengths.		***
ACCESS		· · · · · · · · · · · · · · · · · · ·			МЕ	N X MINUTES
Ну	drauli	c Bay access doors.				
REPLACEMENT PROCE	DURE				ME	N X MINUTES
REPLACEMENT PROCE	DOKE				ME	N X MINUTES
bo of Fi	th end the a t and	and secure the elect: is of the length of co adjacent lengths. secure the length of re with the fixed secu	onduit to	to the aircraft		
741-3437-2-5						

INSPECTI	ON							N	MEN X	MINUTES
			secure length Check	that the early and property and property that the explosion its	operly fit onduit. loop is fi	ted through	cely,			
FUNCTIO	NAL CHECK	(S						M	MEN X	MINUTES
									1	
GROUND	HANDLING	AND GRO	UND TEST	EQUIPMEN	Т					
SPECIAL	TOOLS TO	REMOVE O	R SERVICE							
REMARKS	5									
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SYSTEM						
FIRCTRICAT		SUB-SYSTEM		COMPON	NENT	REF. NO.
ELECTRICAL FIRE PROTECTION Fire Detector Test Swit				Fire Detector	Test Switch	11-12-6
AVRO PART NO. 90S-S-160		MANUFACTURER	MAN'F	"R'S PART NO.	AIRCRAFT E	
OVERHAUL LIFE:	KNOW	/N-	ES	TIMATED- 1500	hours	
FUNCTION		ilitate testing the co or conduits for contin		of the fire		
LOCATION		and Ground Test panel				
ACCESS					ME	N X MINUTE
	Open hi Remove	inged access panel, fo panel E21 by releasin	rward of g ll cam	LH speed brake. loc fasteners.		
REPLACEMENT PR	OCEDURI	 E			ME	N X MINUTE
	and nu	d secure switch to pan t supplied. t and secure circuit w panel - ll camloc fast	ires to		•	i i

INSPECTI	ION					-,77		MEN	X MINUTES
			Operat action	e the swi		eck that	mounted. the lever and break		
FUNCTIO	NAL CHECK	(S						MEN	X MINUTES
GROUND	HANDLING	AND GRO	UND TEST	EQUIPMEN	т				
SPECIAL	TOOLS TO	REMOVE O	R SERVICE						
REMARKS	5								
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SYSTEM						
		SUB-SYSTEM		COMPON		REF. NO.
ELECTRICAL		FIRE PROTECTION	1	Fire Extingu Delay Relays		11-12-7
AVRO PART NO.		MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT	EFFECTIVIT
CS-R-126		Rogers Majestic			25	201
OVERHAUL LIFE:	KNOW	N-	ES	TIMATED- 1500 h	ours	
FUNCTION	which p is disc irrespe One uni	by the extinguisher acceptants the low pressure tharged. Also, complete tive of the length of the stand one in the extinguishment of the extinguishment of the extinguishment.	are fuel etes the of time t ctinguish	cock to close be actuation pulse he fire warning er actuation cir	fore the extended for four second light is deposit for the	tinguisher conds pressed.
LOCATION		ccessory Panel E5, loc Missile Bay.	cated on	bulkhead at stat	ion 485	
ACCESS						IEN X MINUTE
REPLACEMENT PRO	OCEDURE				М	IEN X MINUTES
REPLACEMENT PRO	Fit and	d secure unit to pane t and secure circuit wand position the pane	wiring to	terminals.	М	IEN X MINUTE
REPLACEMENT PRO	Fit and Connect Raise a	d secure unit to pane t and secure circuit wand position the pane	wiring to	terminals.	М	EN X MINUTE

INSPECT	ON						МІ	EN X	MINUTES
		Check	that the that the operly con	circuit wi	ecurely mov	inted. ecurely	EL T		
FUNCTIO	NAL CHECKS						М	EN X	MINUTES
GROUND	HANDLING AND GROU	IND TEST	EQUIPMEN	Т					
SPECIAL	TOOLS TO REMOVE OR	SERVICE							
		*							
REMARKS	5								
ISSUE	1								
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SYSTEM	,	SUB-SYSTEM		COMPON	IENT	REF. NO.
ELECTRICAL		FIRE PROTECTION	NC	Fire Protection Engine Shut-off		11-12-
AVRO PART NO. MS25024-1		MANUFACTURER	MAN'F	'R'S PART NO.	AIRCRAFT E 25201	
OVERHAUL LIFE:	KNOW	V-	ES	TIMATED- 1500 h	nours	
FUNCTION	engine to clos	nergized, by depressing fire warning light, the the relevant LH or as in preparation for	the relay	rs complete a sup pressure fuel cod	ply circuit	te
LOCATION	Panel F	3, located on the RH	wall of	the nose wheel w	vell.	
ACCESS					ME	EN X MINUTE
	Unobstr	ructed.				
REPLACEMENT PRO	OCEDURE	-			ME	N X MINUTE
	mountir	i secure relay in pan ng screws. t and secure circuit				

INSPECTION	MEN X MINUTES
Check that the cir	ay is securely mounted. cuit wiring is securely cted to the relay terminals.
FUNCTIONAL CHECKS	MEN X MINUTES
GROUND HANDLING AND GROUND TEST EQUIPMENT	
SPECIAL TOOLS TO REMOVE OR SERVICE	
REMARKS	
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7#1-3413-2-6

SYSTEM		SUB-SYSTEM		COMPON	JENT	REF. NO.
SISILM		30B-3131EM				IKEI . NO
ELECTRICAL		FIRE PROTECTION		Hydraulic Bay Relay	Lock-Un	11-12-9
AVRO PART NO.		MANUFACTURER	MAN'F	R'S PART NO.	AIRCRAFT	EFFECTIVIT
MS25024-1					2520	01
OVERHAUL LIFE:	KNOW	N-	ES	TIMATED- 500 1	nours	
FUNCTION	light, to the	nergized, by depressi the relay transfers remaining extinguish on for a second shot.	the LH er	ngine extinguishe	er actuation	
LOCATION	Panel I	6, located on the ro	of of nos	se wheel well.		
ACCESS					М	EN X MINUTE
	Thebat	+4				
	Unobati	ructea.				
REPLACEMENT PRO	OCEDURE				М	EN X MINUTE
		d secure relay to pan ng screws.	el using	four		
	Connec	t and secure circuit	wires to	relay terminals		
					_	

INSPECT	ION					MEN >	MINUTES
		Check that	the relay is so the circuit win connected to the	ring is se	curely and		
FUNCTIO	NAL CHECKS					MEN	MINUTES
GROUND	HANDLING AND GRO	DUND TEST EQU	IPMENT				
SPECIAL	TOOLS TO REMOVE C	DR SERVICE					
REMARKS	5						
ISSUE	1						
DATE	4 Feb 57						

SYSTEM		SUB-SYSTEM		COMPO	NENT	REF. NO.
ELECTRICAL		FIRE PROTECTION		Second Shot Switch		11-12-10
AVRO PART NO.	C	MANUFACTURER	MAN'F	FFECTIVITY		
OVERHAUL LIFE:	KNOWN	V-	ES.	TIMATED- 1500 1	nours	
FUNCTION		olete the supply for on second shot circu		nguisher		
LOCATION	Front	cockpit LH console, p	anel El4.	, state		
ACCESS					МЕ	N X MINUTES
	Unobstr	ructed when panel E14 - seven quick-faste	is remov	ved from the		
REPLACEMENT PRO	CEDURE			(-1000-0	МЕ	N X MINUTES
	Connect Fit and	d secure switch and gasher and nut supplie t and secure circuit d secure panel E14 in fasteners.	d. wires to	switch.		

INSPECTI	ON				-			MEN	X MINUTES
			Operat action	ly and pro e the swit is smooth	operly fit	eck that t	he lever		
FUNCTIO	NAL CHECK	(S						MEN	X MINUTES
GROUND	HANDLING	AND GRO	UND TEST	FOLIIPMEN	т				
SPECIAL	TOOLS TO	REMOVE O	R SERVICE						
REMARKS	5								
ISSUE	1								
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FUNCTION Wher compared. LOCATION Panel ACCESS	SUB-SYSTEM FIRE PROTECTION MANUFACTURER The energized, by operating pletes the preparatory secure by the selected first elected first elected first elected.	ESTIMAT g the second econd shot actuat	shot switch, etuation circuit.	AIRCRAFT 252 hours	REF. NO. 11-12-11 EFFECTIVITY 201 EN X MINUTES
AVRO PART NO. MS25024-1 OVERHAUL LIFE: KNO FUNCTION Wher compared to the c	MANUFACTURER DWN- n energized, by operating pletes the preparatory solution by the selected first elected first elected on the root batructed.	ESTIMAT g the second econd shot actuat	PART NO. TED- 1500 shot switch, etuation circuit.	AIRCRAFT 252 hours	EFFECTIVITY 201
MS25024-1 OVERHAUL LIFE: KNO FUNCTION Where compared a set- LOCATION Pane ACCESS Unol	n energized, by operating pletes the preparatory so up by the selected firstell E6, located on the root batructed.	g the second econd shot actuat	shot switch, etuation circuit.	252 hours	201
FUNCTION Wher compared to the	n energized, by operating pletes the preparatory so up by the selected first elected first elected on the root batructed.	g the second econd shot ac t shot actuat	shot switch, ctuation circ tion circuit.	euit	EN X MINUTES
LOCATION Pane ACCESS Unol	pletes the preparatory solution by the selected first elected first elected first elected on the root between the selected on the root between the selected elected el	econd shot actuat	ctuation circuit.	evit	EN X MINUTES
ACCESS	bstructed.	of of nose wh	neel well.	м	EN X MINUTES
Unol				М	EN X MINUTES
REPLACEMENT PROCEDU					
mou	and secure relay on pan nting screws. nect and secure circuit				EN X MINUTES

INSPECTI	ION								MEN X	MINUTES
			Check	that the that the cly fitted	circuit wi	ring is se	curely an	đ		
FUNCTIO	NAL CHECK	(S							MEN X	MINUTES
GROUND	HANDLING	AND GRO	UND TEST	EQUIPMEN	т					
SPECIAL	TOOLS TO	REMOVE O	R SERVICE							
31 23112	10020 10	TEMOTE O	N SERVICE							
DEMARK										
REMARKS	5									
		1								
ISSUE	1									
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SYSTEM		SUB-SYSTEM		COMPON	NENT	REF. NO
ELECTRICAL		FIRE PROTECTION		Inertia Sw	itch	11-12-12
AVRO PART NO. CS-S-150	Mir	MANUFACTURER MAN'F'R'S PART Minneapolis-Honeywell			AIRCRAFT EFFECTIVI	
OVERHAUL LIFE:	KNOWN	1–	ES	TIMATED- 1500	hours	
FUNCTION		o under a deceleration, simultaneously, thus.				
LOCATION	Aft fac	ce, LH side, of bulk	nead at s	tation 485.	. Viii - 1	
ACCESS					ME	N X MINUTE
	Remove	electrical power bay locs.	7 access	door -		
REPLACEMENT PRO	CEDURE				ME	N X MINUTE
	bolts.	d secure unit to bulk t and secure circuit rminal cover - two so	wires to			

INSPECT	ION							MEI	X MINUTES
			Check and pr	that the that the coperly cound then re	circuit wi	res are se	nted. curely		
, 1-61-s									
FUNCTIO	NAL CHECK	(S						ME	X MINUTES
GROUND	HANDLING	AND GRO	UND TEST	EQUIPMEN	Т				
SPECIAL	TOOLS TO	REMOVE O	R SERVICE						
REMARKS	5								
						1 - 2 - 1 - 7			
ISSUE	1								
DATE	4 Feb 57								