

Report # 7-0400-34 Sheet # 1-1 Issue 5 Prepared By: K. Griffin Checked By: E. Burnett

INTRODUCTION

The following is a Weight & C.G. Summary of the C-105 MK 2 Production Aircraft. Of necessity some of the figures quoted herein are only of a preliminary nature, particularly in the case of the Rear Fuselage, Engine Installation and some systems which are being extensively redesigned and about which there is as yet very little information available.

Information has been based on the MK 1 Aircraft with relevant changes made sometimes to preliminary scheme drawings and sometimes verbal information. Where no details are available, allowances have been retained as in the MK 1 Aircraft.

GENERAL:

- a) Orenda PS 13 Engines comprise the Power Plant (4,500 lb each).
- b) A package containing 4 "semi-submerged" Sparrow II Missiles (432 lb each) is currently carried.
- c) The R.C.A. Radar Astra I System is installed -- Weight & C.G. location being to the latest breakdown received from them, dated Jan. 11/57. However, the Sparrow III Auxiliaries included in the above breakdown have been deleted and some AVRO installed Antennae added (Total Weight = 2,679.5 lb.)
- N.B. It should be noted that the first MK 2 Aircraft will probably not be as this summary designates, but will be a Flight Test version with Instrumentation replacing the Missile Package etc.

1. STRUCTURE

a) Wing:

WEIGHT (1b)

13

MAR 18 1957

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I/Wing complete - miscellaneous production drawing changes, inclusion of actual weights of some machinings

There will be changes to the Inner Wing I/B Ribs, Centre Trailing edge etc., associated with the redesigned engine mounts on 31 Research etc., however, since complete details are as yet unavailable, weight allowances will remain as in MK l Aircraft. It should also be noted that actual weights of MK 1 machined skins have been checked and the estimates currently recorded tally with the actual weight at nominal thicknesses. However, some skins, notably those machined at AVRO are in some cases well above maximum tolerances, excess weight being of the order of 6 lb/panel. Thus with a set of 12 similarly machined skins excess weight could be as much as 70 lb/A/C. It is hoped that

> Weight Change Increase Ref. 2-1

13

Fin - redesign elevator jack pick-up, some actual weights etc.

machining techniques will have been improved for the MK 2 A/C

and nominal thicknesses more strictly adhered to

Weight Change Increase

Ref. 2-1

Date: March 1st, 1957 Aircraft: C-105 MK 2



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INTRODUCTION

1. STRUCTURE:

N.B. In the MK 1 Aircraft Structural changes are made to the Wing, Fin and Control Surfaces to accommodate the possible necessity of installing a "Buzz Damping" System. These allowances will not be made on the MK 2 Aircraft until the requirements for Damping are confirmed, pending flight test of MK 1 A/C.

c)	Fuselage Sta. 255"-485"	WEIGHT	(lb)
	Miscellaneous Items - all joint items at Stas. 255 & 485 now treated separately (see below) Equipment Bay Structure and Radar Bay Structure will also change, but estimates are not yet available.	6	13
	Weight Change Decrease Ref. 2-2	G	13
d)	Duct Bay Sta. 485"-591.65"		
	Lower Panel - all joint items at Sta. 591.65" treated separately (see below) The Floating Duct is being extensively redesigned with an articulated adaptor between floating duct and engine to include gills. No drawings are yet available and it is felt that the weight currently recorded for the duct, to-	æ	12
	gether with the adaptor ring (see Engines group) will be adequate. An estimate will be made at the earliest opportunity.		
	Weight Change Decrease Ref. 2-2	Ð	12
e.)	Engine Bay Sta. 591.658-742.58		
	Longerons - joint straps called up on "Marry-up" drawings treated separately (see below) Outer Skins - Aft of Sta. 717.36 gauge decrease to .051 from	0	11
	رَمْلِ. Intermediate Formers - Former 717.36 now intermediate, weight	0	3
	was previously light former Inner Longerons - entirely redesigned - cap from Sta. 591.65-	+	4
	697.28 now steel etc.	+	28
	Engine Access Doors - alterations to surround structure and former tips, fairing added etc. Light Formers - miscellaneous minor changes		5
	Weight Change Increase Ref. 2-2	+	10



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		INTRODUCTION	
1.	STR	UCTURE (continued)	
	f)	Rear Fuselage Sta. 742.5 Aft.	WEIGHT (1b)
		Fixed R.F some changes to MK l production drawing estimates are applicable here, parabrake release socket (6 lb) now with structure (see Equipment Group)	+ 12
		Weight Change Increase Ref. 2-3	+ 12
	g)	Fuselage Joints	
		This is a new sub-section to include all items of "Marry-up" of Fuselage components i.e. items not readily assigned to the above structure components. Items of Fin and Wing "Marry-up" are already included in the structural weights of these groups.	
		Fuselage Marry-Up [*] - not previously considered separately (* Actual Weight change per A/C = 52-11-12-13 = 16 lb/A/C see above structural breakdown).	+ 52
		Weight Change Increase Ref. 2-3	+ 52
		TOTAL STRUCTURAL WEIGHT INCREASE	+ 66
2.	LAN	DING GEAR	
	Mai	n Undercarriage - redesign of shortening mechanism preliminary estimate of Weight Saving by Dowty	50
		TOTAL LANDING GEAR DECREASE Ref. 2-3	- 50
3.	POW.	ER PLANT & SERVICES	
	Fir	ine Controls - based on MK l production drawing estimates e Extinguishing System - MK l production drawing estimates essory Gear Boxes on Fuselage - Heat Exchangers called up here (see also Equipment Group) Some MK l production drawing changes, some oil lines etc. previously omitted	+ 1 - 5 + 21 + 22
	Sta	dix Fuel Control Unit - deleted, replaced by Lucas Units of 13 or 14 lb each installed by Orenda rter Shroud - first scheme drawings available cellaneous changes	- 120 + 6 + 3

OTAL POWER PLANT & SERVICES DECREASE

NSECREASSIFIE

Ref. 2-4



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3. POWER PLANT & SERVICES (Continued)

N. B. Although it is fully appreciated that the MK 2 A/C Fuel System will be entirely different from that of the MK 1 A/C Proportioner System, there are currently insufficient details available to evaluate the weight of the MK 2 A/C Selection System which will replace the Proportioners.

4.	FLYING CONTROLS GROUP	WEIGHT (1b)
	Mechanical Controls - redesign of rudder bearings and bellcranks F/C Hydraulics - Spec. weights were recorded for all servos, now	+ 6
	some actual weights obtainable Manufacturer s Weights of pumps - allowance	+ 20
	too low Filter element retainers now steel	+ 28 + 5
	Actual Weights of accumulators and compensators target weights were	÷ _9
	previously recorded	+ 23
	Miscellaneous alterations.	* 1
	TOTAL FLYING CONTROLS INCREASE	+ 83
	<u>Ref. 2-5</u>	
5.	EQUIPMENT GROUP	
	Oil & Hydraulic Fluid Cooling - Heat Exchangers now with Accessories Gear Boxes (see Engine Group)	s = 22
	Utility Hydraulies - re-issues of MK l production drawings reduction	
	in compensator target weight etc.	- 10
	Emergency Power Pack - Initial allowance too high	- 3
	Radio & Radar - latest R.C.A. weights of Jan. 11/57, Sparrow III Auxiliaries not carried	- 2
	Brake Parachute - MK 1 production drawing estimates, socket now	~
	in structure.	- 7
	Electrics - addition of brake parachute solenoid valve	+ 1 .
	TOTAL EQUIPMENT GROUP DECREASE	- 43
	Ref. 2-6	
6.	OPERATIONAL LOAD	
	Engine Oil Trapped - more detailed estimate in piping etc.	+ 4
	TOTAL OPERATIONAL LOAD INCREASE Ref. 2-6	+ 4





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INTRODUCTION

SUMMARY

Weight Change - Aircraft Weight Empty

Structure + 66 lb.

Landing Gear - 50 lb.

Power Plant & Services - 72 lb.

Flying Controls + 83 lb.

Equipment - 43 lb.

= 16 lb.

Weight Change - Operational Load Less Fuel

Trapped Oil

+ 4 lb.

Weight Change - Operational Weight Empty -(A/C less Fuel)

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44,197 lb. 44,385 lb. = 12 lb.





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WEIGHT CHANGES TO 7-0400-34 Issue 4 of February 1st, 1957

WEIGHT CHANGE	WEIGHT (1b)
Wing:	
+ 3.84 Centre Line Joints I/W = detailed estimate of hard from aux. spar to main spar.	lware
+ 8.40 Ribs I/W - Actual weight of Rib # 4 M/S to C/S fwee section.	ì.
+ 6.04 F/Spar to M/Spar Struct. production drawing esting 2 pt. refuelling door Actual Weights of Fus. Side Rightion bracket transfer pump	= 5.54 lb.Aft. + 10.29
- 9.07 M/Spar to R/Spar Structure - re-estimate of posts on obtained actual weights Error in previous estimate	+ 4.02 - 13.09 - 9.07
- 3.86 Wing to Fus. Joint - production drawing estimate :	
- 1.46 Strut.Pick-Ups I/W - Minor design changes and errocalcs. of engine mtg. pick-up.	or in
+ 8.64 Structure for Main U/C = re=estimate of U/C pivot details of rib # 24 L/Edge - a	
previously An actual weight was obtained of Main U/C pick-up fitting - this	+ 5.74
estimated	
	+ 8.64
+ 12.53	
Fin & Rudder	
+ 4.33 Fin - redesign of jack pick-up Actual Weights of hinge bracket "Marry-Up" re-estimated	+ 1.46 + 0.79 + 2.08
	+ 4.33







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WEIGHT CHANGES TO 7-0400-34 Issue 4 of February 1st, 1957

WEIGHT CHANGE

WEIGHT (1b)

Centre Fuselage Sta. 255"-485"

- 12.50 Miscellaneous Items C.F. - the joints at Sta. 255 & 485 were included here now see separate section on fuselage joints below.

12.50

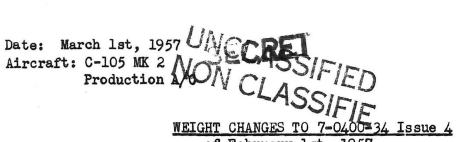
Duct Bay Sta. 485" - 591.65"

- 12.33 Lower Panel - based on MK 1 details - joint to E.B. now separated see below.

12.33

Engine Bay Sta. 591.65"-742.5

0	11.30	Longerons - Joint straps now included with D.B. to E.B. joint - see separate joint section below.		
0	2.99	Outer Skins - Aft of Sta. 717.36 gauge changed from .064 to .051		
+	9.89	Intermediate Formers - Former Sta. 717.36 was previously considered as a light former - now included here since it is considerably heavier than the average light former, although it is not machined.		
	24.75	Light Formers - Former Sta. 717.36 now with Intermediate Formers Light Formers Sta. 596.45-692.45 terminate at Lower Longeron, frame tip box structure now considered with access doors (see below)	0 0	6.09
		Lower Formers 7-2.3-725.7 re-estimated	-	1.08
		Miscellaneous changes splice plates etc	-	1.58
-	77 (0		-	24.75
+	11.60	Engine Access Doors - deletion of MK l type surround structure	6	24.00
		Former tip structure added	+	29.42
		Addition of Fairings	+	6.18
			+	11.60
+	27.73	Inner Longerons E.B From Sta. 697-742.5 structure was		
		assumed as MK 1 - redesign now	+	5.60
		Sta. 591.65-697.28 - Cap Steel was	_	22.13
+	10.18	Inner Longerons E.B From Sta. 697-742.5 structure was assumed as MK 1 - redesign now Sta. 591.65-697.28 - Cap Steel was Al072 straps on lower cap etc.	T	~~ · I)



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of February 1st. 1957

WEIGHT CHANGE

WEIGHT (1b)

Rear Fuselage Sta. 742.5 Aft.

- 12.54 Fairing under Rudder - estimate to MK 1 production drawings, socket for parachute release now included with structure.
- 5.72 Longerons & Torque Box - based on MK 1 production drawing estimates.
- 5.13 Formers Fixed R.F. - based on MK 1 production drawing estimates.
- 11.95

Joints Fuselage:

This is a new sub-section to include all items of "Marry-up" of Fuselage components i.e. items not readily assigned to the above structural components. Items of Fin and Wing Marry-up are already included within the structural weights of those groups. Previously the weight of these Marry-up items was included with one or other of the mating components.

- 5.00 Joint C.F. to D.B. - new report, an allowance previously in Miscellaneous Structure.
- Joint F.F. to C.F. new report, an allowance was previously 8.14 included in Miscellaneous Structure C.F.
- 9.52 Joint Air Intakes to F.F. - was included with Air Intake Structure. This structure is currently being remestimated and it is felt that the joint allowances etc. were inadequate, hence the introduction of this further weight.
- 25.08 Joint D.B. to E.B. - new report, allowances were previously in Lower Panel D.B. and Longerons E.B. (see relevant sections)
- 4.43 Joint R.F. to E.B. - new report - was included in R.F. Weight

Undercarriage Group

Main Undercarriage - redesign of shortening mechanism preliminary estimate of weight ving SSI 50.00 NONSECRESIFIE by Dowty.

^{52.17}

Date: March 1st, 1957 Aircraft: C-105 MK 2

UNICASSIFIED NON CLASSIFIE

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WEIGHT CHANGES TO 7-0400-34 Issue 4 of February 1st, 1957

WEIGHT CHANGE	WEIGHT	(lb)
Power Plant & Services		
	based on production drawing estimates MK 1 g System - based on production drawing estimates MK 1 A/C	
+ 43.08 Accessory Gear Box	xes on Fuselage - addition of heat ex- changers called up here (see also Equipment Group) + 2 Constant Speed drive and engine oil lines etc. based on MK 1 production drawing estimates. + 2 + 4	
- 267.00 Engine Service Acc	cessories - deletion of Bendix fuel con- trol units, replaced by Lucas units of 13 or 14 lb each installed by Orenda. They state that their engine weight will cover this increase, hence - 12 Constant Speed Drive units now with Accessories Gear Boxes - 14	0.00
+ 155.63 Starters & Gear B	oxes on Engine - Constant Speed Drive Unit transferred here (see above) + 14 Starter Shroud - scheme estimate +	7.00 5.66 2.97
- 71.89		
Flying Controls Group		
	ls - redesign of rudder bearings and bellcranks lics - Actual weights of accumulators	
	and compensators - target weights were previously recorded + 2 Increase in servos (based on	2.90 7.00
	Increase in weights of pumps, now manufacturer's quotes, previously only	8.00
*	Miscellaneous changes to production etc +	2.42
+ 6.93 F/C Hydraulies Find UNCLASSIFIED RELASSIFI	n - Servo specification weight increase 4 lb to 9 lb 4 Filter element retainer Steel 5 Miscellaneous changes valves and	5.00 1.09 0.84 6.93
DEC IM	continued.	

Date: March 1st, 1957
Aircraft: C-105 MK 2
Production A/C.

WEIGHT CHANGES TO 7-0400-34 Issue 4

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of February 1st, 1957

WEIGHT CHANGE	WEIGHT (1b)
Flying Controls Group (continued)	
+ 5.07 F/C Hydraulics I/W - Servos actual weights were Spec. weights Miscellaneous, minor changes	+ 4.00 + 1.07 + 5.07
+ 5.28 F/C Hydraulics O/W - Servos actual weights were Spec. weights Filters element retainers Steel Miscellaneous changes & actual weights piping and couplings etc.	+ 4.00 + 3.72 - 2.44 + 5.28
+ 83.22	
Equipment Group.	
- 22.00 Oil & Hydraulic Fluid Cooling - deleted, remaining heat exchangers now incorporated in Pow Plant Group where they will be call up with the Accessories Gear Box istallation	led
+ 1.36 Electrics R.F addition of solenoid valve for brake	
parachute - 7.31 Brake Parachute - based on an estimate to MK 1 drawings,	
socket now included with structure + 2.05 Utility Hydraulics F.F based on MK 1 drawings - actual weights of swivels, addition of valves, piping changes etc.	
- 1.71 Utility Hydraulics C.F based on MK 1 drawings - some piping changes, actual weights of pressure switches, couplings etc.	
+ 4.87 Utility Hydraulics D.B based on MK l drawings - additi of Emergency Power Pack (was inclu	
in Electrics see below)	+ 19.01
Reduction in compensator target weight and fluid Miscellaneous changes piping,	- 10.03
coupling etc.	- 4.11
- 0.88 Utility Hydraulics E.B general revisions to MK 1 pro-	+ 4.87
duction drawings + 4.80 Utility Bydraulics I/W - redesign of swivel assys. UNCLASSIF Enddition of flow control valve	+ 1.70
and rost til ot or	+ 1.80
SECRELASS F Piping re-routed etc. as MK 1	$\frac{+ 1.30}{+ 4.80}$
conti	nued.

Date: March 1st, 1957 Aircraft: C-105 MK 2

Production A/C



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WEIGHT CHANGES TO 7-03-0574 Issue 4 of February 1st, 1957

(Sparrow III auxiliaries not carried)

WEIGHT CHANGE

WEIGHT (1b)

Equipment Group (Continued)

-	22.00	Electrics D.B Emergency Power Pack now in Hydraulics
		see above.
-	3.00	Radio & Radar Fixed - revised R.C.A. figures of Jan 11/57.
-	43.25	Radar Removable - revised R.C.A. figures of Jan. 11/57
		(Sparrow III auxiliaries not carried)
+	44.00	Missile Pack Electronics - revised R.C.A. figures of equip-
		ment mounted within the pack -

ad

+ 3.84 Engine Oil Trapped - more detailed estimate of piping on

MK l A/C - probably similar figures
apply for MK 2

+ 3.84

- 11.57 TOTAL WEIGHT CHANGES





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WEIGHT AND C.G. SUMMARY

P. INCORT DO TON	WEIGHT	H. ARM	V. ARM	
DESCRIPTION	1b.	ins.	ins.	
STRUCTURE	18,277.80	565.94	137.70	
Wing	9,962.04	643.15	142.16	
Fin & Rudder	1,020.70	754.06	209.27	
Fuselage Fwd. Sta. 255"	2,415.63	186.17	129.37	
Sta. 255#-485#	1,672.59	379.71	129.51	
Sta. 485#-591.65#	949.54	533.12	103.76	
Sta. 591.65"-742.5"	1,513.14	661.62	109.24	
Sta. 742.5" Aft.	691.99	795.69	128.15	
Marry-up	52.17	470.59	103.77	
UNDERCARRIAGE - Retracted	2,554.33	487.53	134.52	
Main Undercarriage	1,901.62	539.31	141.00	
Main U/C Doors & Fairings	294.36	539.29	136.01	
Nose Undercarriage	333,81	170.80	99.70	
Nose U/C Door & Fairing	24.54	162.24	88.23	
POWER PLANT & SERVICES	10,774.34	673.94	121.51	
Engines & Accessories PS 13	9,090.12	690.98	121.17	
Gear Box & Drives on Fuselage	281.84	601.70	102.49	
Engine Controls	32.43	375.76	118.61	
Gear Box, Starter & Drive on Engine	315.45	609.12	105.24	
Engine Nose Bullet	70.00	587.17	116.00	
Fire Extinguishing System	70.46	702.44	134.83	
Engine Mountings	156.53	666.00	136.69	
Fuel System	757.51	543.10	135.69	
FLYING CONTROLS GROUP	1,793.94	686.84	140.05	
Mechanical Flying Controls	942.92	688.24	148.66	
Hydraulic Flying Controls	851.02	685.29	130.52	
EQUIPMENT FIXED & REMOVABLE	7,948.78	325.31	111.61	
Instruments	46.07	163.68	138.70	
Probe	23.00	9.74	108.00	
Cockpit Pressure Sealing	5.00	186.00	130.00	
Oxygen System	43.44	227.71	142.18	
Air Conditioning System	897.00	331.77	133.38	•
Hydraulic Main System	597.49	498.76	116.66	
Cabin Insulation	11.91	179.24	130.00	
Brake Parachute	62.38	769.41	143.24	
Electrical System	1,216.85	427.55	112.61	
Low Pressure Pneumatics	39.01	478.47	127.28	
Surface Finish	100.00	591.52	140.20	
Intake De-icing Boots	51.84	197.02	118.00	
Radome Anti-icing	8.88	51.49	125.00	
Canopy Actuation	62.05	222.11	154.47	
Cabin Consoles	17.45	174.76	124.34	
Radar Door Actuation	10.00	268.00	95.00	
Ejector Seats	186.00	201.10	136.25	
Radio & Radar removable	1.906.80	178.38	104.17	
Radio & Radar fixed	55 F 12.00	234.67	115.71	
TOUTO OF TOUGHT TITLE IN THE PARTY OF THE PA		1000		
	850.00	390.84	96.00	
Sparrow Pack Structure Sparrow Pack Mechanisms	850.00 625.32	390.84 376 . 67	96.00 99.22	

Date: March 1st, 1957 Aircraft: 105 MK 2

Aircraft: 105 MK 2
Production A/C



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DESCRIPTION		WEIGHT	H. ARM	V. ARM	M.A.C.
<u>DESCRIPTION</u>		<u>lb.</u>	<u>ins.</u>	<u>ins.</u>	_\$
Equipment (Fixed & Remov.) (Cont	(d.)				
Sparrow Pack Hydraulics		350.19	368.83	99.00	
Sparrow Pack Electronics		143.00	332.00	100.00	
Sparrow Pack Electrics		64.10	362.29	95.00	
AIRCRAFT WEIGHT EMPTY		41,349.19	548.23	128.37	
USEFUL LOAD		18,625.76	516.38	136.67	
Crew		430.00	194.00	136.50	
0 il		138.97	636.92	110.57	
Alcohol		22.00	93.00	138.00	
Engine Fire Extinguisher Fluid		25.00	730.00	129.00	
Residual Fuel		218.40	553.98	134.04	
Missiles		1,728.00	389.29	88.30	
Oxygen Charge		13.39	259.69	159.91	
Water for Air Conditioning		260.00	268.00	132.00	
Fuel for Combat Mission	77 /0 97	15,790.00	542.05	142.30	
Normal Combat Mission	U/C Up	EO 081 0E	538.34	130.95	28.27
Normal Compat Wission	U/C Down	59,974.95	520 O/	300 03	0.0 F23
Half Combat Mission Fuel	U/ U DOWIA	7 00 5 00	539.94	128.31	28.71
(1,012 gals. @ 7.8 lb/gal.)		7,895.00	543.20	139.55	
(1,012 gals, 6 /10 10/gals)	U/C Up		527 OF	128.81	20 10
Combat Weight (Half Combat	dy o op	52,079.95	537.95	120.01	28.17
Mission Fuel)	U/C Down	J~,01707J	539.79	125.77	28.67
	U/C Up		537.01	126.89	
Operational Weight Empty	o/o op	44,184.95	997.01	120.09	27.91
obergorous werens numbel	U/C Down	ر ده بست و بسب	539.18	123.31	28.50
	U/C Up		543.02	128.46	29.56
Operational Weight Empty	oyo op	42,456.95	743.02	120,40	29.50
Less Missiles	U/C Down	4~9470077	545.28	124.74	30,19
Max. Internal Fuel (2,492 gals.		19,438.00	541.85	144.16	
@ 7.8 lb/gal.		17,430,00	741.07	144.10	
	U/C Up		538.49	132.16	28.31
A.U.W. Max. Internal Fuel		63,622.95			
the state of the s	U/C Down		540.00	129.67	28.73
Max. External Fuel (500 gal.		4,226.00	522.34	60.64	
@ 7.8 lb/gal. + Drop Tank)					
	U/C Up		537.48	127.71	28.04
A.U.W. Max. Internal &	-12	67,848.95	22. U-q-		
External Fuel	U/C Down		538.89	125.38	28.42
			1 2 1		

N.B. 1) Aircraft Datum = 120 above arbitrarily chosen ground line.

³⁾ Normal Mission Fuel requirements have been increased in the order of 600 lb by Aerodynamics Dept.



²⁾ Revised fuel tank capacities in Centre Fuselage Tanks for MK 2 A/C are now incorporated.

			U	NSF	CRI				R.	TFO	Ma.	7-0)	\ - 3	4-5	
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