INCLASSIFIED

UNCLASSIFIED

AVRO AIRCRAFT LIMITED INTER-DEPARTMENTAL MEMORANDUM

Ref. 5049/11/J

Date January 1st, 1958

To See Distribution

From Mr. E. F. Burnett

Subject ARROW 2 PRODUCTION A/C WEIGHT & C.G. SUMMARY - Report # 7-0400-34 Iss. 15

Attached is a copy of Weight and C.G. Summary Report #7-0400-34 Iss. 15, dated January 1st, 1958 for your retention.

This report is revised monthly and is issued complete on the lst of each month.

Classification cancelled / changed to: UNCLASSSIFIED

By authority of: DRDA 7/DARFT 5-8/DAS Eng 6-4-5

Date: 5 Nov 1992

Signature: B. aulycy

Unit / Rank / Appointment: DSIS 3, Secretary CRAD HQ DRP

amc.

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Report # 7-0486-24 Iss. 15

Date: January 1st, 1958
Aircraft: ARROW 2
PRODUCTION A/C

INDEX

Sheet #	Content
1-1 to 1-4	Introductory notes and explanation of Weight Changes.
2-1 to 2-2	Weight & C.G. Summaries
3	Horizontal C.G. plot showing basic fixed points on flight envelope. The variation of C.G. with fuel used has been omitted until such time as a fully approved fuel sequencing is established.
4-1 to 4-15	I.B.M. Detail Sheets of Weights and C.G.'s.

Date: January 1st, 1958
Aircraft: ARROW 2
Production A/C

Report # 7-0400-34 Iss. 15 Sheet # 140N CLASSIFIE

INTRODUCTION : WEIGHT CHANGES

The following is a Weight & C.G. Summary of the Arrow 2 Production Aircraft, based on the latest weight estimates available.

All Weight & J.G. changes are relative to Issue 14 of December 1st, 1957.

The early Arrow 2 Aircraft # 25206 etc. will not be as this summary designates, but will be flight test aircraft with Instrumentation, "Astra Minus" Radar, Missile Pack Trial Installations, various equipment T.I.'s (e.g. Ram Air Turnine in Aircraft # 25206 and 25211) etc. Weight statements for these aircraft will be issued at a later date.

GENERAL:

- a) Orenda PS 13 Engines comprise the Power Plant (4,500 lb each, brochure weight, excluding Nose Bullet and Input Frame).
- b) A package containing 4 "semi-submerged" Sparrow II Missiles (432 lb each) forms the current armament.
- c) The R.G.A. Astra I Radar System is installed, Weight & C.G. breakdown being to the latest information received from R.G.A. dated 4th October 1957. No Sparrow III Auxiliaries are carried in the production aircraft. Some Avro installed antennae have been added. The Infra-red Tracker System is allowed for in its proposed location at the Fin. (Total Weight = 2,881.8 lb)

In addition to the above the Minneapolis-Honeywell MH 64 Damping System is installed.

Arrow 2 Aircraft # 25206 to 25208 will have the "Astra Minus" System (Navigational & Communication equipment only).

d) A considerable number of Arrow 1 parts, assemblies and equipment have now been weighed. Where these weights apply to the Arrow 2 Aircraft, they have been recorded in the Arrow 2 records.

On the I.B.M. Sheets in this report, immediately preceeding the item title, will be found a number varying from 0 to 100. This is the percentage actual weight recorded in that report.

1. STRUCTURE

WEIGHT (1b)

a) Wing:

Ribs # 1 to # 9 I/Wing - redesign of Arrow 2 fabricated
ribs, reductions in caps & Webs,
also minor redesign in machined
ribs - 25.86

Spars I/Wing - production drawing estimates of C/Spar Aft - 0.37

F/Spar to M/Spar Struct I/W - redesign 2 point refueling
door and gate valve operating
mechanism + 3.27

Date: January 1st, 1958
Aircraft: ARROW 2
Production A/C

SESTARET Report # 7-0400-34 Iss. 15 Sheet # 1-200 CLASSIFIE

INTRODUCTION & WEIGHT CHANGES

	INTRODUCTIO	ii d whitairi oimidab		
1. STR	UCTURE		WEIGH	TT (1b)
<u>a)</u>	Wing: (Cont'd.)			
	of En	production drawing estimates gine Mountings		
	fitti Elevator Control Box - Actua	1 Weights of some applicable	+	9.22
		1 parts	+	1.21
	weigh	t Decrease Wing		L2.53
b)	Fin & Rudder:			
	No We	ight Change		
c)	Front Fuselage Fwd. Sta. 255	п		
	Cockpit Formers - Production Crew's Bulkheads - first Arr		-	2.70
	estimates		-	0.32
		nnel supports at U/C Uplocks	+	0.55
	Pilot's Canopy - addition of		+	0.54
	Radar Nose Structure - Produ	Fuselage Weight Decrease		0.71
	FIGHT	ruserage weight Decrease		0.17
d)	Centre Fuselage Sta. 255"-48	<u>5"</u>		
	Equipment Bay Structure - re	vision of estimate of Mounting		
		ay	+	0.17
		initial Arrow 2 scheme draw-		-
		estimates	*	5.25
	Cent	re Fuselage Weight Increase	+	5.42
e)	Duct Bay Sta. 485"-591.65"			
	No W	eight Change		
f)	Engine Bay Sta. 591.65-742.5	<u>u</u> :		
	No W	eight Change		
g)	Rear Fuselage Sta. 742.5" Af	t.		
	etc. n and pr Outer .025 to			1.71
		m to N155 etc. etc. UNCLASSIFIED /	NON	CLASSIFIÉ

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	INTRODUCTION & Weight CHANGES	CLASSINE
1.	STRUCTURE	WEIGHT (1b)
	g) Rear Fuselage Sta. 742.5" Aft. (Contid.) Removable Tailcones - Contid Inner Skins - some gauge changes	↑ 11.08
	h) Fuselage "Marry-Up"	- 12(6)/
	No Weight Change.	
	TOTAL STRUCTURAL INCREASE	+ 119.57
2.	LANDING GEAR	
	No Weight Change	
3.	POWER PLANT & SERVICES	
	No Weight Change	
4.	FLYING CONTROLS GROUP	
	Flying Control Hydraulics I/Wing - addition of surge dapmers to Elevator jacks (mod. to jack) Flying Controls Hydraulics O/W - addition of surge dampers to Aileron jacks. Flying Controls Hydraulics Fin - addition of surge damper to Rudder jack	1.501.500.75
	FLYING CONTROLS GROUP-INCREASE	+ 3.75
5.	EQUIPMENT GROUP	
	Electrics Air Intakes - Arrow 2 Production Drawing estimates Electrics D.B Westinghouse proposal for Transformer- rectifier unit greater than previously re-	- 1,90
	corded Lucas Rotax Unit	+ 9.00

Date: January 1st, 1958

Aircraft: ARROW 2

Production A/C

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Sheet # 1-4 CLASSIFIE

INTRODUCTION & WEIGHT CHANGES

5.	EQUIPMENT GROUP	WE	IGHT (1b)
	Electrics E.B Westinghouse proposal for Alternator		
	Wt. = 87 lb, Lucas Rotax was 82 lb Utility Hydraulics I/Wing - Dowty increase in weight of	+	10.00
	Main U/C Door jacks	+	2.00
	Radio & Radar Fixed Equipment - latest R.C.A. estimate Radio & Radar Removable Equipment - latest R.C.A. estimate, increase due primarily, so they state, to the fwd. nose estimate now backed by 85% Actual	+	54.00
	Weight.	4	103.80
	Missile Auxiliaries in Pack - latest R.C.A. estimate Electrics & Electronics fixed in Pack - first complete	-	12.50
	estimate made for production pack, partially based on test pack information. (additional to R.C.A. equipment.)		
	Actuation System	+	4.20
	Firing System - not previously included Missile Pack Mechanisms - addition of nylon liners in	+	85.20
	telescopic links	+	2.77
	Modifications to door actuation levers First detailed estimate of all installation	+	0.51
	details Production drawing estimates of launchers (approximately 61.5 lb each, previously 75 lb	+	19.65
	was allowed.)	-	54.38
	EQUIPMENT GROUP INCREASE	4	222.35

SUMMARY

Weight Change - Aircraft Basic Weight

Structure + 119.57 lb
Flying Controls+ 3.75 lb
Equipment + 222.35 lb
+ 345.67 lb

Weight Change - Operational Weight Empty (A/C less Fuel)

<u>Issue 14</u>
<u>Issue 15</u>
<u>45,263.82 1b</u>
<u>45,609.49 1b</u> + <u>345,67 1b</u>

Date: January 1st, 1958 Aircraft: ARROW 2

Production A/C

UNCLASSES R E T Report #17-0400-34 Iss. 15 Sheet # 241 CLASS/F/F

WEIGHT & C.G. SUMMARY

	WEIGHT	H. ARM	V. ARM
DESCRIPTION	1b	ins。	ins.
	Manufacture (Manufacture (Manuf	Will Charles Annual Charles Company	
STRUCTURE	19,044.15	566.37	137.35
Wings	10,012.03	642.54	1142.22
Fin & Rudder	1,024,00	754.25	209.28
Fuselage - fwd. Sta. 255"	2,572.87	183.01	128.43
Sta. 255"-485"	1,691.83	375.38	130.60
Sta. 485"-591.65"	1,119.04	538.28	105.41
Sta. 591.65"-742.5"	1,550.20	661.04	110.51
Sta. 742.5 Aft.	1,021.85	805.37	128.62
"Marry=Up"	52.33	467.60	103.84
LANDING GEAR - RETRACTED	2,542.31	487.09	134.78
Main Landing Gear	1,901.62	539.31	141.00
Main Gear Doors & Fairings	282.34	537.60	138.37
Nose Landing Gear	333.81	170.80	99.70
Nose Gear Door & Fairing	24.54	162.24	88.23
POWER PLANT & SERVICES	10,800,67	671.41	121.32
Engines & Accessories PS 13	9,186.78	687.95	121.16
Gear Box & Drives on Fuselage	281.84	601.70	102.49
Engine Controls	32.43	375.76	118.61
Gear Box, Starter & drives on Engine	315.45	615.98	105.24
Engine Nose Bullet (Orenda Supplied)	70.00	587.17	116.00
Fire Extinguishing System	65.46	700.45	134.27
Engine Mountings	132.38	666.82	136.52
Fuel System	716.33	530.91	134.48
FLYING CONTROLS GROUP	1.6792.80	683.95	140.43
Mechanical Flying Controls	944.53	684.69	148.25
Hydraulic Flying Controls	848.27	683.13	131.72
EQUIPMENT FIXED & REMOVABLE	8,640,80	335.93	115.30
Instruments	46.07	163.68	138.70
Probe	15.00	=38.14	108.00
Cockpit Pressure Sealing	5.00	186.00	130.00
Oxygen System	26.07	240.54	156.70
Ejector Seats	339.06	204.43	134.10
Air Conditioning System	856.00		
	61.0.81	333.49	134.98
Hydraulics Main System	640.81	504.22	117.77
Cockpit Insulation	14.31	187.48	132.00
Drag Chute	91.07	786.68	143.19
Electrical System	1,276.26	434.44	112.66
Low Pressure Pneumatics	56.94	421.96	128.47
Surface Finish	100.00	591.52	140.20
Intake De-icing Boots	88.00	195.82	118.00
Canopy Actuation	64.92	221.99	154.35
Cabin Consoles	17.38	174.63	124.32
Radar Door Actuation	10.00	268.00	95.00
MH64 Damping System	180.15	471.27	135.69
Radio & Radar Removable	2,011.30	213.06	114.24
Radio & Radar Fixed	685.00	236.96	118.63
Sparrow Pack Structure	890.19	381.67	97.60
Sparrow Pack Mechanisms	551.12	374.40	99.12
Sparrow Pack Hydraulics	331.94	373.52	99.00
Sparrow Pack Electronics Removable	185.50	331.91	100.00
Sparrow Pack Electrics & Electronics fixed Sparrow Pack Air Conditioning UNCLASSIFIED	153.50	340.74	95.00

Date: January 1st, 1957 Aircraft: ARROW 2

Production A/C

SEVEE T Report of 7-0400-34 Iss. 15 Sheet # 240N CLASSIFIE

WEIGHT & C.G. SUMMARY

DESCRIPTION	WEIGHT 1b	H. ARM	V. ARM ins.	M.A.C.
U/C Up		546.58	128.83	
AIRCRAFT BASIC WEIGHT	42,820.73			
U/C Down		548.81	125.13	
USEFUL LOAD (less fuel)	2,788.76	377.69	104.73	
Crew	390.00	194.00	136.50	
Oil	138.97	636.92	110.57	
Engine Fire Extinguishing Fluid	25.00	730.00	129.00	
Residual Fuel	218.40	553.98	134.04	
Missiles	1,728.00	389.29	88.30	
Oxygen Charge	13.39	259.68	159.91	
Water for Air Conditioning	275.00	268.00	131.75	
U/C Up		536.25	127.36	27.70
Operational Weight Empty	45,609.49			
U/C Down	479007847	538.35	123.89	28.28
U/C Up		542.04	128.89	29.29
	43,881.49	542.04	120.07	27.627
Operational Weighing Empty Less Missiles U/C Down	45,001,49	544.22	705 08	29.89
Less Missiles U/C Down		544.22	125.28	27.09
Normal Combat Mission Fuel *	77 070 00			
(2,227 gal. @ 7.8 lb/gal.)	17,370.00	80	-	as
		-	-	-
Normal Combat Weight	62,979.49			
		0	CC	C
Half Combat Mission Fuel *				
(1,113 gal. @ 7.8 lb/gal.)	8,685.00	-	€	ED .
Combat Weight * (half				
Mission Fuel)	54, 294.49			620
Maximum Internal Fuel				
(2,492 gal. @ 7.8 lb/gal.)	19,438.00	541.85	144.16	
U/C Up		537.92	132.38	28.16
A.U.W. Maximum Internal Fuel	65,047.49	121072	-52050	20025
U/C Down	->9-4104	539.39	129.95	28.56
Maximum External Fuel		757657	1-10/)	20070
(500 gal. @ 7.8 lb/gal + Drop Tank)	4,248.00	521.84	60.67	
U/C Up	4,240,00			07 99
	60 00T 10	536.93	127.98	27.88
A.U.W. Maximum Internal and	69,295.49	F20 23	700 50	00 01
External Fuel U/C Down		538.31	125.70	28.26

- N.B. 1) Aircraft Datum = 120" above an arbitrarily chosen ground line.
 - 2) * Fuel weights in accordance with latest data issued by Aerodynamics November 19th, 1957. Centres of Gravity have temporarily been omitted until a fuel sequencing system is finally established.
 - 3) With the increase of 344.40 lb in Operational Weight Empty the Normal Combat Mission Fuel has increased by 90 lb.

